

AMERICAN ART MUSEUM | NORTHEAST UNITED STATES

FINAL REPORT

ADVISOR: HEATHER SUSTERIC

April 3, 2013

SEAN FELTON | STRUCTURAL

AMERICAN ART MUSEUM | NORTHEAST, UNITED STATES

PROJECT TEAM

Owner Representative : Gardiner & Theobald General Contractor : Turner

Engineers/Consultants

Civil: Philip Habib & Associates
Structural: Robert Silman Associates
MEP: Jaros, Baum & Bolles
Lighting: ARUP



N ELEVATION RENDERING

STRUCTURE

- Foundation consists of drilled caissons under pile caps, 36" concrete secant wall, and 24" pressure slab
- Composite floor system 3.25" concrete slab and W-shape beams
- Lateral system works with steel braced frames and specified rigid floors
- Floors supported by combination of columns, trusses, and hangers
- 30' cantilever at level 5 (SE corner)
- Levels 3 and 4 hung from level 5 in several places

MECHANICAL

- 5 architecturally exposed cooling towers
- Mechanical space in cellar, level 2, level 4, and level 9
- Combination of VAV for galleries and CAV for less controlled spaces
- Roof heating/snow melting roof system

LIGHTING/ELECTRICAL

- Lamps and windows specified for optimal color rendering (CRI > 97)
- LEDs, fluorescents
- (4) 4000 A 208Y/120V switchboards serve building

Architects

Executive: Cooper, Robertson & Partners

Design: Renzo Piano Building Work Shop

GENERAL INFORMATION

Function: Museum/Mixed-Use

 Size:
 220,000 SF

 Height:
 150 ft

Number of Stories: 9 above, 2 below Construction: 5/2011—12/2014 Cost: \$266 million

Delivery Method: Design-Bid-Build (GMP)



SE CORNER RENDERING

ARCHITECTURE

- Façade and interior module of 3'-4"
- Stainless steel and precast concrete cladding
- Exposed structural steel and MEP systems
- 50,000 SF of interior gallery space
- Step-backs for outdoor gallery space
- Skylight and architectural fabric used in level 8 gallery

CONSTRUCTION

- Secant wall poured in tandem with excavation
- 42" steel tubes used to stabilize secant wall during construction
- Deep wells gather site water before desedimentation, pumped back into sewer system

http://www.engr.psu.edu/ae/thesis/portfolios/2013/shf5014/index.html



ACKNOWLEDGEMENTS

I especially thank the following people and organizations for their assistance, provision, guidance and advice, support, prayers, patience, flexibility, humor, distractions, generosity, consideration, games of Dutch Blitz and Chess, and for otherwise picking up my slack over the past year.

Companies The Owner (who wished to remain anonymous)

Turner Construction Barton Malow

Cooper, Robertson & Partners

RPBW

Robert Silman Associates

Advisors and Faculty Professor Bob Holland

Professor M. Kevin Parfitt

Heather Sustersic

Dr. Thomas Boothby Dr. Andres Lepage

Classmates Sarah Bednarcik

Dan Bodde
Eric Buckwalter
Jim Chavanic
Tyler Donnell
Jon Fischer
Jon Gallis
Adam Karlheim
Nick Leonard
Cheuk Tsang
Andrew Voorhees

Family Dad & Phyllis, Brandi

Mom, Liz, Christina

Roommates Bobby Frederico & James Wheeler

Friends Carter Bowman

Emily Draving
Zach Jones
Ian Kelmartin
Kevin Moyer
Abby Ott

Meredith Tipton

EXECUTIVE SUMMARY

The Final Report investigates the possibility of supporting the South-Eastern corner of Renzo Piano's American Art Museum (AAM) without the use of a column at 3-M.5 (circled in Figure 1 below). Due to the monumental nature of the project, the structural alterations would need to be done in a way that minimized impacts on the architecture of the building. Though it was understood at the outset of this investigation that the weight and cost of the structural system would almost certainly increase, these effects were also to be minimized. After a thorough design and investigation of the proposed structural system and its effects on the architecture and construction of the building, this report recommends that the current structural design by Robert Silman Associates is the best solution to supporting AAM's signature cantilever.

A load path was successfully developed that did not involve a column at 3-M.5. This load path requires the use of additional trusses along the East wall of the Main Gallery space and South wall of the office spaces on Levels 3 and 4. Special consideration was taken to ensure that exposed structural steel in the gallery aligns with the carefully-developed modular façade system established by the Architect, and that the sizes of these trusses and their members did not affect the exterior envelope of AAM.

This concern for the architecture, however, adversely affects the weight and cost of the building. The proposed changes increase the weight of the influenced structural system by 50%, or nearly 100 t. Also, the foundations require greater capacity and 5 additional piles of varying strengths. In all, the cost of the structural system would increase by nearly \$2 million, or 33%. Additional provisions isolated to individual members and custom cross sections will greatly increase the difficulty of the construction.

Note: All photos and renderings used with permission from RPBW.



Figure 1: Renderings of current (top-right) and proposed (bottom-right) structural systems.

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BUILDING INTRODUCTION

The American Art Museum (AAM) will serve as a replacement to the owner's current facility in the same city. Figure 2 shows AAM's new location in a more vibrant of the district city where aging warehouses, distribution centers, and food processing plants are being renovated and replaced by art galleries, shops, and offices. Now AAM stands in place of several such warehouses, and will provide a magnificent new southern boundary to the city's recently renovated elevated park, which terminates on the eastern edge of the site.

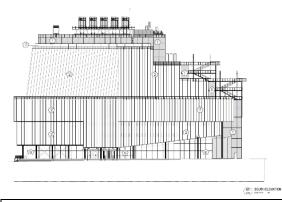


Figure 2: Arial map showing urban location along river (www.maps.google.com)

Renzo Piano's approach to AAM's design and architecture serves to reference the city's history with large cooling towers and outdoor terraces that step back towards the river on the west. These outdoor terraces will provide views into the city and space for outdoor exhibits and tall sculptures while being protected from any wind by the higher portions of the building's west side. Alternately, the large cantilevers, insets, large open spaces, exposed steel, and modular steel plate cladding show no attempt to camouflage AAM with the more historical surrounding buildings.

AAM's façade is comprised of the aforementioned stainless steel panels, pre-cast concrete, and glazing using a standard module of 3'-4" (about 1m; shown in Figure 3). The steel panels, the primary element of the façade, are 2 modules wide, or 6'-8". While most of the façade components are broken at each story, the longest panels stretch 60' on the southern wall from levels 2 to 6 and from 6 to 9.

This new facility is a multi-use building with gallery and administration space, two café/restaurants, art preservation and restoration, a library, and a 170-seat theater. Public space including the theater, classrooms, restaurants, and galleries are located on the south half of the building on the ground level and levels 5 through 8.



administration are dispersed on the north side at each level. The 220,000 square-foot AAM will stand 158' tall and has a guaranteed maximum price of approximately \$267 million. Construction began in May 2011 and is expected to be complete in December 2014.

Mechanical, storage, conservation, offices, and

Figure 3: South Elevation showing modular façade (A-007)

EXISTING STRUCTURAL SYSTEMS

OVERVIEW

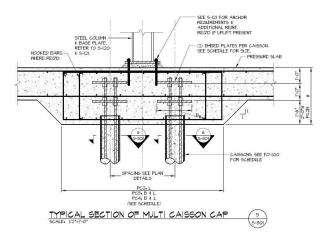
AAM sits on driven steel piles filled with reinforced concrete with diameters of either 9.875" or 13.375" and grouped by pile caps. From the foundation level at 32' below grade, 10 levels rise on steel columns and trusses. Each floor is designed for steel/concrete composite bending. The lateral system consists primarily of braced frames spanning several stories. At some levels however, the floor system uses HSS diagonal bracing between joists and beams to create a rigid diaphragm that also transfers the lateral loads between staggered bracing. Moment frames are used for localized stability purposes. While masonry is used in AAM it is used for fire rating purposes only.

The building classifies as Occupancy Category III. This is consistent with descriptions of "buildings where more than 300 people congregate in one area" and "buildings with a capacity greater than 500 for adult education facilities."

FOUNDATIONS

URS Corporation published the geotechnical report in February 2011 to summarize the findings of several tests and studies performed between 2008 and 2010. They summarize that while much of the site is within the boundaries of original shoreline, a portion of the western side is situated on fill-in from construction. They explain further that the portion that was formerly river has a lower bedrock elevation and higher groundwater. Due to the presence of organic soils and deep bedrock, URS suggested designing a deep foundation system and provided lateral response tests of 13.375" diameter piles reinforced with 3"-diameter bars and socketed into bedrock.

The engineers acted on the above suggestions and others. The piles are specified with a 13.375" diameter of varying concrete fill and reinforcement to provide different strengths to remain consistent with URS Corp's lateral response tests. Low-capacity piles (9.875" diameter) are individually embedded to the pressure slab, while typical and high-capacity caissons are placed in pile caps consisting of one or two caissons. The high-capacity caissons are always found in pairs and are located beneath areas of high live load or where cantilevers are supported. For a complete layout and caisson schedule, see FO-100 in Appendix A.



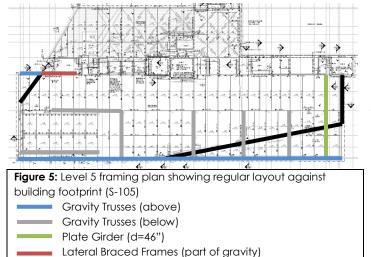
A pressure slab and the perimeter secant-pile walls operate in tandem to hold back the soil and groundwater below grade during construction and for the lifespan of the building. The walls vary between 24" and 36" and are set on 6'-6" wall footers and caissons. These are isolated from the pressure slab shown in Figure 4. Hydrostatic uplift led the engineers to design a 24" pressure slab, isolated from the 5" architectural slab-ongrade by a 19" layer of gravel.

Figure 4: Pile cap section (S-301)

GRAVITY SYSTEM

FLOOR SYSTEM

A surprisingly regular floor layout contrasts the obscure geometry of the building (Figure 5). The engineers managed to create a grid with spacings of roughly 20' (E-W) and 30' (N-S), where the 20' sections are divided by joists which support the floor decking running E-W. Beams that do not align with the typical perpendicular grid indicate a change of building geometry below or above. Each joist and beam/girder is designed for composite bending with the floor slab.



Four slab/decking thicknesses are called for depending on deck span and loading, all on 3"-18 gauge composite metal deck. The most common callout is 6.25 (total thickness) lightweight concrete. This provides a 2-hour fire rating. 7.5N (normal weight) is used on level 1 for outdoor assembly spaces and the loading dock, and 9N is used for the theater floor. The roof above the level 9 mechanical space calls out 5.5.

While the layout can be considered relatively consistent, the beam sizes and spans selected suggest a much more complicated floor system. Though a typical span at 20'-30', spans often run as

long as 70' on the gallery floors (levels 6-8). The shorter spans require joists as small as W14x26, but the longer spans supporting the upper gallery levels require beams as large as W40x297s for web openings. In several places welded plate girders are specified at depths from 32.5" to 72." The plate girders are used as transfer large loads and moments over cantilevers, especially from gravity trusses and lateral braced frames (Figure 6).

FRAMING SYSTEM

Outline of Building Below

Cantilevers on the south side of AAM are supported by 1 or 2-story trusses, typically running in the N-S direction. One large gravity truss runs along the southernmost column line between levels 5 and 6 to support the cantilever on the south-eastern corner of the building.

While the vast majority of columns are W12x or W14x shapes, some of the architecturally exposed steel vertical members are HSS shapes, pipes, or solid bars. Furthermore, the gravity load path goes up vertically and horizontally nearly as much as it flows directly down a column to the foundation. Figure 7 shows how large portions of the southern half of AAM's levels 3 and 4 are hung from trusses and beams on the level 5 framing system.

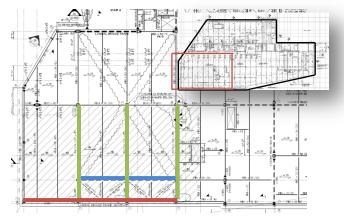


Figure 6: Level 3 framing plan showing transfer girders and lateral braced frames (S-103)

Lateral Braced Frame (above)

Lateral Braced Frame (below)

Plate Girder (d=72")

Renzo Piano's designs often expose structural steel, providing an extra constraint on the design team. One example is Column 3-M.5 which supports level 5 from the outdoor plaza below. The foundation column below grade specifies a W14x311, a typical shape for a column, but the architecturally exposed structural steel is called out as 22" diameter solid bar. A unique analysis would be required for a solid bar acting as a column, as AISC XIII does not have provisions for such a selection in its tables or specifications. Strength calculations for the optional 22" Round HSS are discussed in the Proposed Structural Design section of the Final Report.

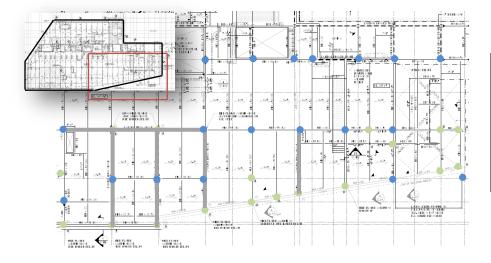


Figure 7: Level 3 framing plan showing hangers and outline of hung/cantilevered portion of building (S-103)

- Gravity Truss (above)
 Compression Support
 (single below)
- Tension Support (single above)

LATERAL SYSTEM

AAM's lateral system is more easily understood than its gravity systems. The concentric braced frames stagger up the building, transferring lateral loads via diagonal bracing within the floor diaphragms on level 3 for the southern portion and 5 for the northern portion as shown in Figure 8. Most of the braced frames terminate at ground level, but three extend all the way down to the lowest level. The bracing members are comprised mostly of W10x, 12x, or 14x shapes in X-braces or diagonals. There are, however, HSS shapes are used with chevron-braces. An enlarged floor framing plan showing the braced frames at level 5 is provided in Figure 9 below.

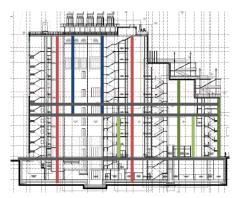


Figure 8: Section cut showing N-S braced frames at staggered heights (A-212)





DESIGN CODES & STANDARDS

The design codes listed for compliance of structural design can be inferred from drawing S-200.01 and Specification Section 014100.2.B:

- International Code Council, 2007 edition with local amendments including:
 - o Building Code
 - o Fire Code
- ASCE 7-05: Minimum Design Loads for Buildings and other Structures
- ACI 318 -08: Building Code Requirements for Structural Concrete (LRFD)
- AISC XIII: Specifications for Structural Steel Buildings (LRFD)
- AWS D1.1: American Welding Society Code for Welding in Building Construction

Other codes not applicable to the structural systems of the building can be found in the specifications.

MATERIALS SPECIFICATIONS

The different materials specifications are summarized in Figure 10 below. Additional information can be found on drawing S-200.0, provided 1 in Appendix A.

	Concrete & Reinforcemer	ıt	Struct	ural Steel		
						Fy
Wt	Use	f'c (psi)	Shape	ASTM	Gr.	(ksi)
LW	Floor Slabs (typ)	4000	Wide Flange	A992	-	50
	Foundations (walls,		Hollow Structural	A500	В	46
NW	slab, pile caps, grade beams)	5000	Structural Pipe	A500	В	46
NW	Composite Column Alternate	8000	Channels	A36	1	36
NW	Other	5000	Angles	A36	-	36
Gr.	Use	ASTM	Plates	A36	-	36
70	Reinforcement	A185	Plates (for Girders)	A709	50	50
150	Reinforcement In Composite Members	-	Connection Bolts	A325-SC	-	80
70	Welded Wire Fabric	A185	(3/4") Anchor Bolts	F1554	36	36
	Fi	gure 10: Mc	aterial specifications			

DESIGN LOADS SUMMARY

GRAVITY LOADS

LIVE LOADS

Perhaps the most notable aspect of AAM's design is its live loads. Typically, one would expect to see Live Loads calculated from ASCE 7-05 minimums (ASCE 7-05 Table 4-1). The structural narrative explains that much of AAM does not fit with any ASCE 7-05 descriptions of use types, so the engineers have provided their own design loads summarized in Figure 11. Additionally the engineers created a live load plan on S-200.01 which shows areas of equal live load on each floor.

The engineers, in a desire for maximum flexibility of the gallery spaces, elected to drastically overdesign the AAM-specific spaces for live loads, while being consistent with ASCE 7-05 minimums for more common areas.

Design Narrative Sum	mary		ASCE 7 Designation
	Live	Live	
Use	Load	Load	Description
Gallery - Typical	100	100	Assembly Area - Typical
Gallery - Level 5	200	100	Assembly Area - Typical
Testing Platform	200	150	Stage Floors
Offices	50	50	Offices
Private	60	n/a	n/a
Assembly/Museum Use Auditorium - Movable Seating	100	100	Theater - Moveable Seats
Compact Storage	300	250	Storage Warehouse - Heavy
Art Handling & Storage	150	125	Storage Warehouse - Light
Outdoor Plaza and Loading Dock	600	250	Vehicular Driveways
Stairs and Corridors	100	100	Stairs and Exit Ways
Lobby and Dining	100	100	Assembly Area - Lobby
Mech Spaces Levels 2, 9	150	n/a	n/a
Mech Spaces Cellar	200	n/a	n/a
Roof - Typical	22 + S	20	Roof - Flat
Roof - Above Gallery	122 + S	n/a	n/a
Figure 11: Comparison	n between	Design l	LL and ASCE 7 Minimum LL

Dead Loads

Because the live loads are so high, special care seems to have been taken by the design engineers to be very precise in their dead load calculations. Similar to the live loads, the diversity of different use types and load requirements have led to a congruent variety of dead load arrangements in structural steel weight, concrete density, MEP requirements, partitions, pavers, roofing, and other finishes. A total of 37 different dead load requirements, arranged by use and location, are listed in the Dead Load Schedule on drawing S-200.01. These range from 76 PSF to 214 PSF. In all, the building has a dead weight of 23,084 k (11,500 tons) from level 1 through level 9 Roof North. Complete dead load calculations for the building are in Appendix B.

Snow Loads

Snow loads were calculated using the procedure outlined in ASCE 7-05. Figure 12 details the summary of this procedure, comparing the Snow Load Parameters on drawing S-200.01 to the City Building Code/ASCE 7-05.

ASCE 7-05 equation 7-1 (section 7.3) states that where the ground snow load exceeds 20 PSF, the flat roof load value must not be less than (20)Is. 22 PSF, the design flat roof load, is not in accordance with ASCE 7's minimum according to equation 7-1 of 23 PSF. It is important to note that the stepback terraces where drifting is a concern are designed for 100-200 PSF of live load, and it is unlikely that the building will experience snow loads exceeding those live loads. Complete | Figure 12: Snow loads comparison Calculations can be found in Appendix B.

Des	sign			
Paran	neters	ASCE 7-05		
Pg	25	25		
Ct	1	1		
Is	1.15	1.15		
Ce	1	1		
Pf	20.1	20.1		
20 Is	22	23		
Figure 12: Snow loads comparison				

Lateral Loads

OVERVIEW

It was not possible to replicate the wind or seismic loads used to design AAM. With greater resources and experience, the engineers used Wind Tunnel Testing and Modal Response Spectrum Analysis as permitted under ASCE 7-05 for wind and seismic loads respectively. These processes allowed the engineers to accurately assess the lateral loading conditions using the correct geometry.

The Final Report does include an investigation of the wind and seismic loads as prescribed by ASCE 7-05. For simplification purposes, only levels 6 (elev. 88' 2") through RN (elev. 169' 10") were considered in this investigation. A series of additional simplifying assumptions allowed for an analysis using ASCE 7-05 chapter 6 for wind and chapters 11 and 12 for seismic. Although the designers determined that seismic loads controlled both base shear and overturning moment in their analyses, The N-S wind case controls base shear and seismic controls overturning in ASCE 7-05 using the Analytical Procedure for wind and Equivalent Lateral Force Procedure for seismic.

Wind Loads

As mentioned above, the wind loads in both directions were found using Analytical Procedure (Method 2) in ASCE 7-05 chapter 6 using assumptions that simplify the geometry and environment of the building. Using the factors in Figure 14 below (calculations in Appendix B), the wind pressures were calculated between 45 PSF and 55 PSF (Figure 15). The design professionals explained that Wind Tunnel Testing returned values of between 30 PSF and 45 PSF, making the Analytical Procedure about 12PSF conservative (a difference of about 20% - 25%).

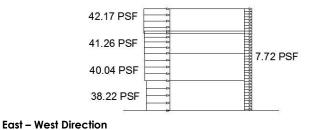
Figure 15 below summarize the revised wind load calculations. The base shears and overturning moments were found for both the North-South (Y) and East-West (X) directions by creating equivalent lateral forces at each story level. More detailed calculations provided in Appendix B show that AAM must resist wind across a much greater surface area in the N-S direction than the E-W. This difference leads to the much greater base shear (1300k which controls) and overturning moment in the N-S direction.

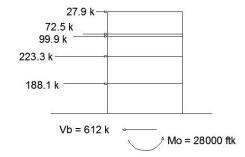
Wind Factors					
E-W N-S					
G _f =	0.89	0.85			
GC pi =	0.55	-			
Cp =	-0.3	-0.5			
Kd =	0.85	1			
Kzt =	1.0	-			
l =	1.15	-			

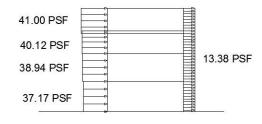
Figure 14 (Left):
Wind factors for
ASCE 7-05
calculations

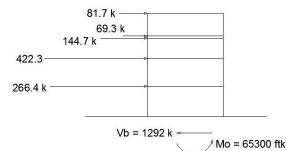
Figure 15 (Below):
ASCE 7-05 Wind











North -South Direction

SEISMIC LOADS

The seismic loads in the Final Report were calculated using the Equivalent Lateral Force Procedure found in ASCE 7-05 chapters 11 and 12. As mentioned above, this method is in contrast to the structural engineer's Modal Response Spectrum Analysis, which is considered to have a higher degree of accuracy (ELF is more conservative). The investigation performed for the Final Report, however, uses the assumptions provided on drawing S-200.01. Figure 16 shows which values were provided by the engineers and which were supplements needed to complete the ASCE 7-05 analysis.

These values were used alongside the revised dead load calculations to find the equivalent lateral forces, base shear, and overturning moment summarized in Figure 17 below. Further calculations can be found in Appendix B. The revised base shear was found to be 1276k for floors 6-RN, much higher than the provided base shear of 946 for the whole building, which can be explained by the different procedures. The overturning moment of 158,500 ft-k controls for both wind and seismic analysis.

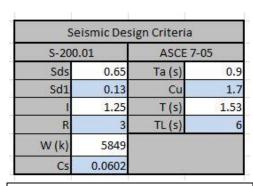


Figure 16: Seismic Design Criteria

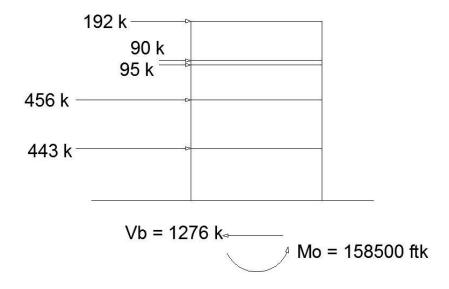


Figure 17: Equivalent Lateral Force Procedure Summary

PROBLEM STATEMENT

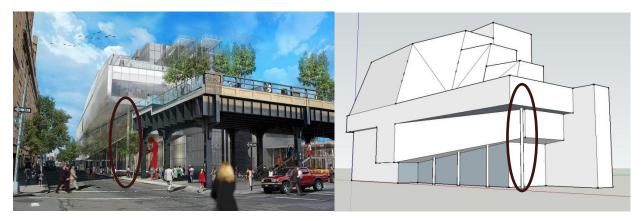


Figure 18: Rendering and Sketchup model showing column 3-M.5 from SE corner

Figure 18 above shows the geometry of AAM at the SE corner entrance and plaza space. Four architecturally exposed columns in the space run parallel to the street and coincide with the horizontal grid of AAM. Three of these columns support the mass of levels 3 and 4 above the glass-enclosed lobby. The fourth column (3-M.5, circled), however, appears to be the sole support of level 5.

A scenario has arisen in which the architect has expressed interest in removing Column 3-M.5. Architecturally, this 22" circular column carries the most delicately-balanced and most massive part of the building visible from street level. Though current design represents an effective and elegant solution to the stability of the cantilever, the architect has asked the structural engineer to consider a method which does not include the use of a column at the location of 3-M.5.

PROBLEM SOLUTION

It is for the above reasons that this thesis project will explore the possibility of supporting the level 5 cantilever without the use of a column at the location of 3-M.5. Extensive changes must be made to the building's gravity load path in ways which minimize effects on the cost, construction schedule, and architectural themes already in place.

A new load path must be introduced to redistribute the 1,800 kips carried by Column 3-M.5. This new load path will require changes to the framing of the levels below and at the cantilever level. First, a two-story truss will have to be added along the south wall (non-orthogonal) on levels 3 and 4 to act as the last support at the cantilever in both directions. Secondly, a truss must be added between levels 5 and 6 at the eastern gallery wall (currently glass). Loads will then travel through the existing frame (where possible), which will be re-analyzed to accommodate the extra loads resisted by each member.

This alternative design will be compared to the current design by analyzing changes to cost, weight, schedule, and impacts on the architecture. Finally, the data will be reviewed by the architect and owner for consideration.

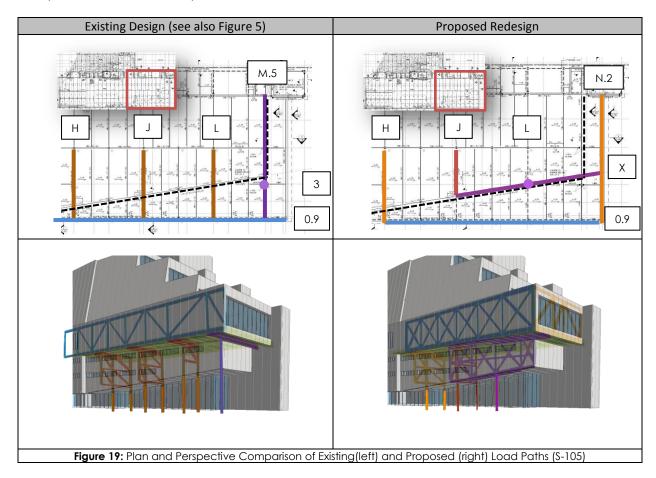
PROPOSED STRUCTURAL DESIGN

LOAD PATH OVERVIEW

Before any technical design could be completed, a load path had to be established. The selection of the proposed load path follows the existing load path as closely as possible in an effort to avoid significant impacts on the architecture in place. Figure 19 shows both the existing and redesigned load paths in plan and perspective.

Floors 5 and 6 are supported by Truss 0.9 on the southernmost edge of AAM. Truss 0.9 is then simply supported, spanning between a strengthened truss at column line H and a new truss at column line N.2. Limitations discussed below in the Final Truss Design section resulted in the design of a cantilever system for Truss N.2 where Truss 0.9 is supported 26' from the nearest support at column line X. A new column was added at the location 6-N.2 to resist uplift. In order to support Truss N.2 at column line X, an additional new truss was designed along the existing exterior face. Truss X was similarly designed as a cantilevered truss supported at column lines L and J. The compression support at L is 42' from its load point due to Truss N.2, and the uplift support utilizes an existing truss at J. Finally the existing Truss J was redesigned to resist that uplift, and existing Truss L was replaced with a column at the location 3-L.

It is important to note that the cantilever supported by 3-M.5 extends 24' beyond its last support and the proposed cantilever extends 46' to its last support at 3-L. Also, for the purposes of this investigation, this alteration to the gravity system has been designed to be entirely independent of the lateral system, and therefore does not impact the rigidity of the structure or any component of the lateral system.



CUSTOM CROSS SECTIONS

EXISTING CUSTOM SECTIONS

As discussed in the Existing Structural Systems section above, AAM's design engineers developed 10 custom shapes to accommodate the large forces and moments created by the cantilevers. Using the information provided on drawing S-201, the strength of each shape was calculated according to AISC XIV and ACI 318-11 in an effort to utilize these custom members designed by the engineers. Furthermore, the designs provided set precedence and guidelines for the development of additional custom members where existing designs are inadequate. The complete calculations for these design strengths are provided in Appendix C.

BUILT-UP PLATE GIRDERS

Though the plate girders provided are used primarily to resist large moments (see Figures 5, 6 and 19, Existing Design), an initial investigation was performed to find both the moment and axial strengths of the plate girders based on a 20' un-braced length. This information was intended for use as a starting point should the large forces in the proposed systems require such capacities. A summary is provided in Figure 20 below.

	Moment Capacity					Axial	Capacity	
Member	φMnx (ft-k)	Lp (in)	Lp (ft)	Limit State	KL/r	KL/r lim	φPn (k)	Limit State
32.5	12197	473	39	Yielding	19.4	113	8395	Torsion
33-1	12518	479	40	Yielding	19.2	113	9446	Torsion
44-1	20520	609	51	Yielding	14.7	113	9532	Torsion
46-1	12555	648	54	Yielding	SL	SL	SL	SL
46-2	29550	657	55	Yielding	13.7	113	16775	Torsion
46-3	22170	631	53	Yielding	14.1	113	9724	Torsion
72-1	45090	815	68	Yielding	10.7	113	10174	Torsion
Figure	20: Plate Gird	der Mome	nt and Co	ompression Stre	engths Assu	uming 20' l	Jn-braced	Lengths

Because each member was found to be compact for flexure, moment capacities are based on plastic section moduli which include both the flanges and web of each member. Plate girder shape PG72-1 has the highest moment resisting capacity of over 45,000 ft-k and a maximum unbraced length of nearly 70'. For compression, however, the web of shape PG46-2 proved to be slender, so it is not considered an option as a component of the proposed truss systems. Shape PG46-2 has the highest compressive strength of 16,775k failing in torsional buckling.

COMPOSITE HSS ROUND COLUMNS

In contrast to the plate girder shapes, the three HSS Round columns function are primarily designed for axial loads. Provisions specified in AISC XIV chapter I2.2 and I3.4 on composite members were used to calculate the compressive, tensile, and flexural capacities of each member, summarized in Figure 21 below. Similarly to the plate girder sections, the strengths provided in Figure 21 are used as a reference if the proposed redesign should require such strengths.

Provisions for composite sections also exist in ACI 318-11 chapter 10. While slenderness checks performed under this specification verified the non-slenderness of the sections, it was decided that the provisions in AISC XIV chapter I should govern the strength design of these members. AISC XIV Equation I2-9b is used to calculate the compressive strength of steel sections without

slender elements filled with concrete. The equation uses the material properties of the concrete and reinforcement without regard for any critical loads, making the reduced stiffness provisions that could be required for slender sections in ACI 318-11 irrelevant to this strength investigation.

Shape	Lu	φMn	φPn	φTn
15A	25	750	2421	2295
15B	25	624	2161	1685
22	45	1714	4389	3545

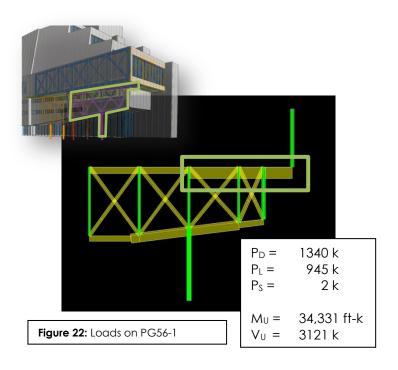
Figure 21: HSS Round Column Capacities

PROPOSED CUSTOM SECTIONS

Although the custom sections provided by the engineers are sufficient for the existing design, the design proposed in the Final Report render all existing cross sections inadequate for the largest required loads. In two locations new custom sections were developed to provide adequate strength for the proposed structural system. Complete design calculations for both proposed custom sections can be found in Appendix C.

PG56-1

Proposed plate girder PG56-1 is designed to transfer loads between Truss N.2 and Truss X (shown in Figure 22). This cross section was developed because of architectural constraints (discussed in the Architecture Considerations section below) which do not allow Truss X to extend past gridline M.5, and limit the depth of the cross section to 56". This depth constraint led to a departure from the component plate dimensions made precedent by the engineers. The web thickness and the flange dimensions were increased to provide additional combined capacity for loadina conditions when used in Truss X. Final design dimensions and capacities for PG56-1 are provided in Figure 23 Below.



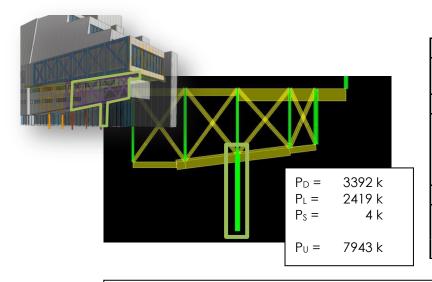
PG56-1 has the largest web thickness, the widest flange width, and the largest flange thickness of any established or proposed plate-girder cross sections. Though PG56-1 is designed adequately for the loads, this departure from precedent component plate dimensions could lead to adverse effects during fabrication and construction. These effects are explored further in the Construction Management Considerations section of the Final Report.

	Existing Shapes		PG56-1		Capacities			
Lb	n/a		20	ft	φMn	41571	ft-k	
D	n/a		56	in	φVn	3402	k	
В	18, 20	in	24	in	φTn	25245	k	
tf	2, 4, 8	in	10	in	φPn	27541	k	
tw	tw 1, 2 in 2.25 in							
	Figure 23: PG56-1 Design Summary							

24R-1

Proposed custom section 24R-1 is designed for column location 3-L (shown in Figure 24), which is the last support for the cantilever in the proposed structural system. Because the loads applied to this column under the proposed system are so high, the current custom round column shapes are inadequate (see Figure 21 above). The proposed section is designed as a composite column using the same conditions and assumptions as the current sections described above.

Figure 24 also summarizes the design dimensions, properties, and capacities of shape 24R-1. In order to acquire sufficient axial strength, the precedent outer diameter of 22" was abandoned for 24", the wall thickness was increased from 1-1/4" to 1-3/4", the concrete strength was increased from 5,000 psi to 15,000 psi, and 14 no. 11 rebars were added for a total of 16. The yield strength of 150 ksi for the reinforcement in the composite columns is not altered from the current design, and can be found on the Custom Round Column Schedule on drawing S-120.01. Though compressive capacity was paramount to the design of 24R-1, the sizing of elements (such as the wall thickness) of the section were developed for fabrication and constructability. These considerations are discussed in the Construction Management Considerations section of the Final Report.



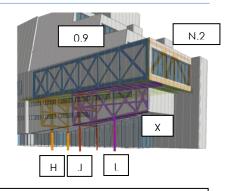
Pipe					
Do	24	in			
t	1.75	in			
	Concrete				
f'c	15000	psi			
fy	150	ksi			
no.	11				
n	16				
	Capacity				
φPn	8272	k			
φTn	8053	k			
φMn	2754	ft-k			

Figure 24: Loads and Capacities of 24R-1

TRUSS DESIGN AND ANALYSIS

OVERVIEW

The proposed truss system was designed primarily using an itterative process in ETABS. Due to the complex nature of this structural system, initial sizes were selected based first on precedence for Truss 0.9, and later judgment as the design progressed down the load path. An analytical method for selecting initial sizes was performed for a variation of Truss X used to verify ETABS's truss action (seen in the ETABS Verification section of this report below), but was not performed for other trusses due to the verification method's



Fiaure 25: Truss Name Summarv

dependence on structural determinacy. Because the overall deflection at the 68' cantilever would be relatively large, X-braces were used where possible to provide extra stiffness and minimize deflections. This provision rendered each truss, with the exception of Truss N.2, statically indeterminate and did not allow for the use of an analytical method for selecting initial member shapes.

BASIC LOADING AND MODELING ASSUMPTIONS

Each truss was modeled independently with simple supports and major-axis moment releases for the diagonal and vertical members. Horizontal members, however, were modeled continuously except where different horizontal cross sections meet. Modeling each truss independently ensured that these simple end-releases are reliable and accurate assumptions, mitigating the effects of out-of-plane effects (i.e. torsion, minor-axis bending) from other steps of the load path at the connnection sites.

IBC 2009 LRFD load combinations found in section 1605.2.1were used to determine the design loads of the proposed system. Equation 16-2 (below) was found to control in all cases for the gravity investigation.

1.4(D + F)	(Equation 16-1)
$1.2(D + F + T) + 1.6(L + H) + 0.5(L_r \text{ or } S \text{ or } R)$	(Equation 16-2)
$1.2D + 1.6(L_r \text{ or S or R}) + (f_1L \text{ or } 0.8W)$	(Equation 16-3)
$1.2D + 1.6W + f_1L + 0.5(L_r \text{ or } S \text{ or } R)$	(Equation 16-4)
$1.2D + 1.0E + f_1L + f_2S$	(Equation 16-5)
0.9D + 1.6W + 1.6H	(Equation 16-6)
0.9D + 1.0E + 1.6H	(Equation 16-7)

Loads were calculated using the Dead and Live Load Schedules found on drawing S-200.01 and applied to the Trusses appropriately. In the case of Truss 0.9 loads had to be calculated from level 6 to the roof level using tributary areas of each member supported by the truss. An additional dead load was added at the locations where columns from upper floors load Truss 0.9. A 2k point load was applied for each level supported by a column. Also, the steel panel exterior wall was estimated to have a weight of 15PSF, and was applied at typical loading points which (see Final Truss Design section below). Once modeled, the reaction from each of the Dead, Live, and Snow loads was used to load the next truss down the load path.

Additionally, the trusses are modeled such that connections are concentric. Diagonal and vertical members utilize only W14x shapes, while the top and bottom chord members use shapes determined to be efficient for both axial and bending forces. For design purposes, ETABS considers the top flanges horizontal members to be fully braced, and diagonal and vertical members to be fully un-braced if constraints are not added explicitly. Due to the preliminary nature of this investigation, P-Delta effects were not considered.

Finally, tension members were considered for yield strength only and rupture will need to be considered when designing the connections.

DEFLECTION CONSIDERATIONS

Due to the nature of this investigation, the proposed trusses have been designed for strength. Deflection was considered for overall deflections at cantilevers and mid-spans in order to verify a serviceable design. The steel design analysis in ETABS considers certain deflection criteria when interpreting the adequacy of a given member which could not be modified. It is for those reasons that deflection failures of individual members were ignored in ETABS and overall deflections were checked for serviceability. Design deflection results and further discussion can be found in the Overall Deflection of the Cantilever section of the Final Report below.

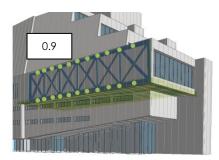
FINAL TRUSS DESIGN

Truss 0.9

Since the floor systems of neither level 5 nor level 6 needed altering for the proposed load path, Truss 0.9 is the primary support for those levels (Figure 26). The truss is loaded along the top and bottom chords. In order to accurately model the combined member loading (bending and axial forces), Truss 0.9 is loaded every 10' according to the beam spacing on the two levels. This adherence to the existing floor framing system loads the truss at each major panel joint and the mid-spans of the top and bottom chords.

Truss 0.9 is supported by Trusses H and N.2 and was modeled using the conditions shown in Figure 27 below. Truss H lies between levels 3 and 5 and therefore supports Truss 0.9 at the bottom chord only. Alternately, Truss N.2 lies between levels 5 and 6 and supports Truss 0.9 at both levels. A "roller" connection was modeled at each level to provide an accurate reaction scenario. Major-axis moment releases are shown. The W14x120 between the roller supports is merely a placeholder and was not considered in the design of the truss.

Member sizes were finalized on criteria of combined loading efficiency, weight, and constructability. Proposed Truss 0.9 weighs 12.2 t more than the current truss (41.2 t) for a total of 53.4 t. More detailed weight calculations can be found in Appendix F.



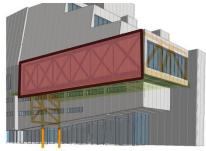
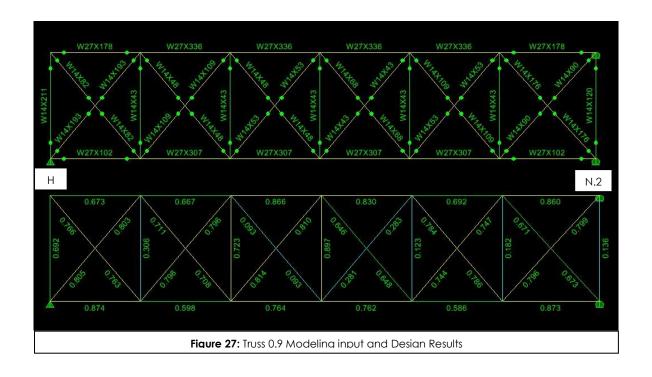


Figure 26: Truss 0.9 Location and Load Path Orientation



TRUSS N.2

As the second step of AAM's proposed load path, Truss N.2 supports Truss 0.9, and cantilevers 26' past Truss X as seen in Figure 28. A tension support is added at column line 6 to resist uplift. Because Truss N.2 runs parallel to the floor framing beams, floor loads at the Eastern edge of levels 5 and 6 are applied as distributed loads. Point loads are applied at column line 0.9 according to the reactions from Truss 0.9. More detailed load calculations can be found in Appendix E.

The shape of Truss N.2 was determined by architectural constraints discussed in the Architecture Considerations section of this report, and was modeled according to the conditions shown in Figure 29 below.

Both the top and bottom chords of Truss N.2 are to be continuous sections for the entire 70' length. The truss weighs 36 t, 9.8 t heavier than the original floor framing. A more detailed weight comparison can be found in Appendix H.

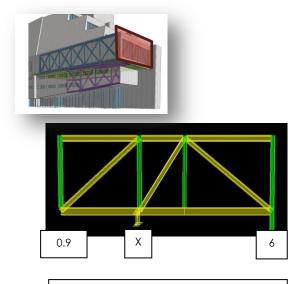


Figure 28: Truss N.2 Location and Load Path Orientation

The selection of Column 6-N.2 is discussed in the Impact on Foundations section below.

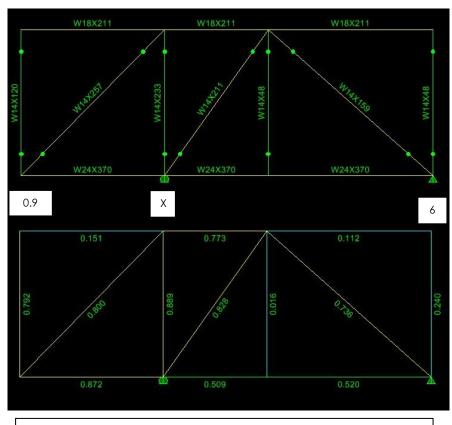


Figure 29: Truss N.2 Modeling input and Design Results

TRUSS X

At over 120 t (see Appendix H), Truss X is the heaviest system of the proposed structural design. Loads from levels 5-9 are applied where Truss X supports Truss N.2 above, and is cantilevered 45' from Column 3-L (see Figure 30). Uplift is resisted by a final truss at column line J. A small distributed load was applied to the top chord at level 5, and point loads were applied at the columns on levels 3 and 4. This placement ensured an accurate model while avoiding unwanted loads applied to the diagonal members. The W16s inserted at level 4 act only as bracing for the diagonal members.

As is further explained in the Architecture Considerations section of this report, the existing architectural envelope limited the depth of Truss X to 56". In order to transfer the loads between the load point at column line N.2 and Truss X,

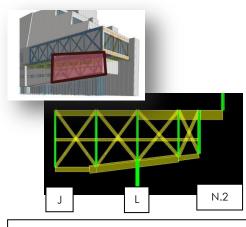
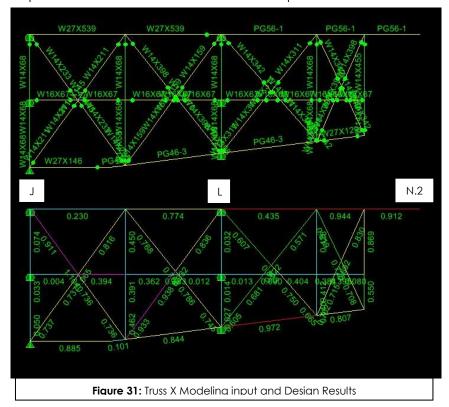


Figure 30: Truss X Location and Load Path Orientation

restricted by the envelope at column line M.5, custom section PG56-1 was designed for adequate shear and moment capacity, and is explained in further detail in Custom Cross Sections section above.

In addition to having the highest weight, Truss X is the only truss system which contains members designed for over 90% efficiency, which can be seen in Figure 31, below. Both the top and bottom chords were deemed acceptable in order to minimize truss weight. PG56-1 weighs 1909 plf, and PG46-3 weighs 748plf (see Custom Cross Sections section above), so an increase in beam size was not considered once a passible capacity was determined. Members in red in the figure which are less than 95% efficient signify a deflection failure.

The vertical member at location X-L was sized as a W14x257, and the design of Column 3-L is also discussed in the Proposed Custom Sections section of this report.



Sean Felton | Structural Option | Advisor: Susteric | April 3, 2013

TRUSS J

The primary purpose of Truss J is to resist uplift caused by the cantilevered Truss X. Figure 32 shows that Truss J is also cantilevered over a support at Column 3-J, which resists uplift, while Column 4-J resists compression. The design of Truss J's supporting columns can be found in the Impact on Foundations section of this report.

Because proposed Truss X spans two stories between levels 3 and 5, it was decided that its uplift support, Truss J, should also cover both stories. Also, for reasons specified in the Architecture Considerations section of this report, the position and orientation of the diagonal members was maintained.

Weighing 45 t in the current designed, the weight of Truss J could be reduced by 36 t (to 9 t) under the proposed system. Figure 33 below shows that the majority of members are W14x68s, the heaviest being a W14x145.

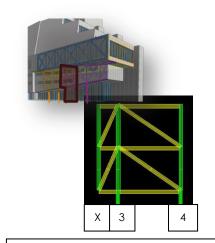


Figure 32: Truss J Location and Load Path Orientation

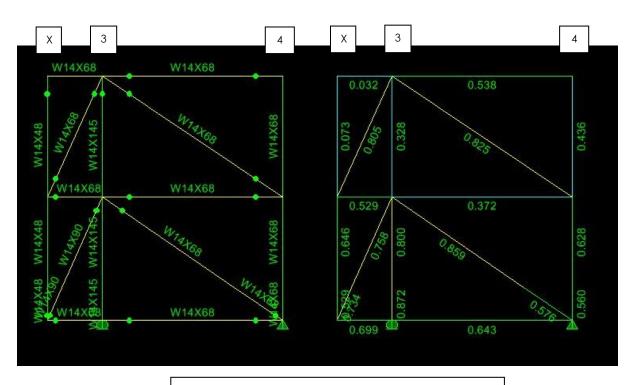
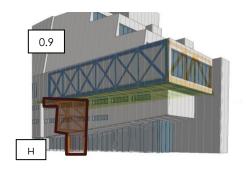


Figure 33: Truss J Modeling Input and Design Results

Truss H

Truss H is the Eastern support for Truss 0.9 (shown in Figure 34). The architectural envelope, further discussed in the Architectural Considerations section of this report, dictate that a single beam must be cantilevered 7' from the rest of Truss H to support Truss 0.9 in a similar fashion to Truss X (see Figures 31 above, 35 below). Furthermore, the first panel of Truss H is cantilevered 12'-6" from its last support at column line 3. Also, red members in Figure 35 below signify failure by deflections.



Similar to Truss J, the shape of Truss H has not changed from the current design for reasons also discussed in the Figure 34: Truss H Location and Path

Architecture Considerations section of this Report. Loads from the Truss 0.9 above and the floor loads were reevaluated and new members were selected.

Truss H features a W14x665 at the location 3-H, the heaviest rolled Wide Flange section in the proposed structural system. Also, custom section PG46-2 was found to be adequate to carry the loads of Truss 0.9 at the top chord.

The design of the supporting columns is discussed in the Impact on Foundations section of this report.

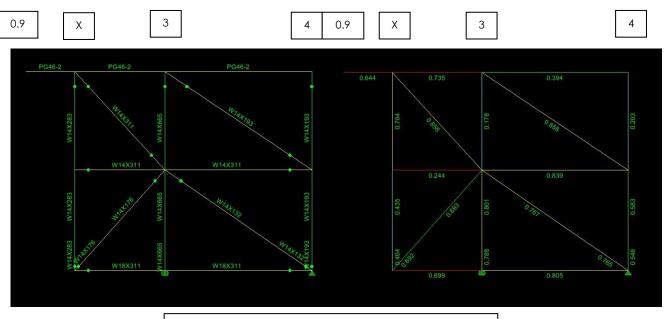


Figure 35: Truss H Modeling Input and Design Results

ETABS VERIFICATION

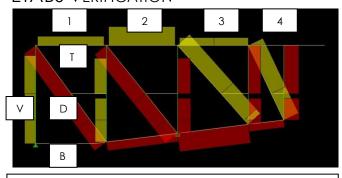


Figure 36: Nomenclature for Truss X Variation

ETABS's analysis of truss shapes was verified with hand calculations using a simplified variation of Truss X shown in Figure 36. First, hand calculations were performed to find the axial loads in each member. Trial member sizes were then selected for ETABS, and the resulting axial forces in ETABS were compared against the axial capacities of the members selected by the hand calculations. A summary is provided in Figure 37 below.

A comparison of loads to capacities was deemed to be more accurate and to better reflect efficiency because it more closely resembles the design process than a comparison of loads alone. Values highlighted green in Figure 27 reflect loads that are conservative compared to the hand values (load exceeds capacity), and the values highlighted in red reflect non-conservative ETABS loads. All load magnitudes, however, are within 10% of the selected member capacities, and therefore verify ETABS's truss analysis.

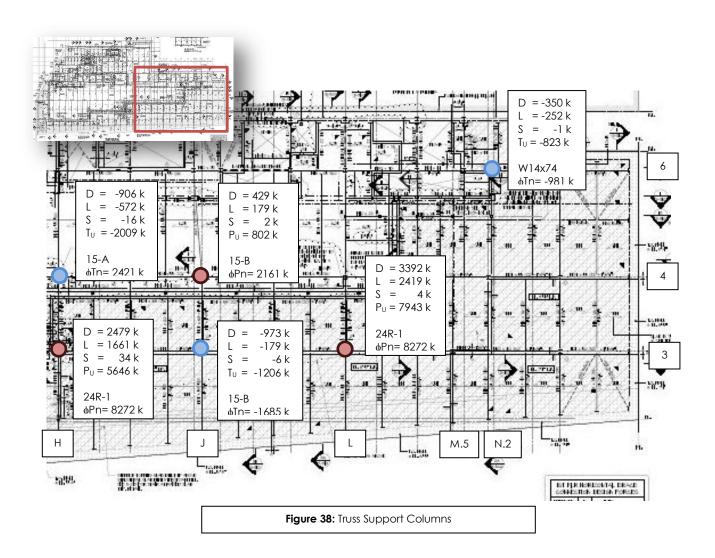
The load patterns used for the hand calculations match those used for the ETABS verification of this model but reflect an earlier iteration of the design process and do not match the loads used for the final proposed design of Truss X. More detailed calculations and selected member sizes can be found in Appendix D.

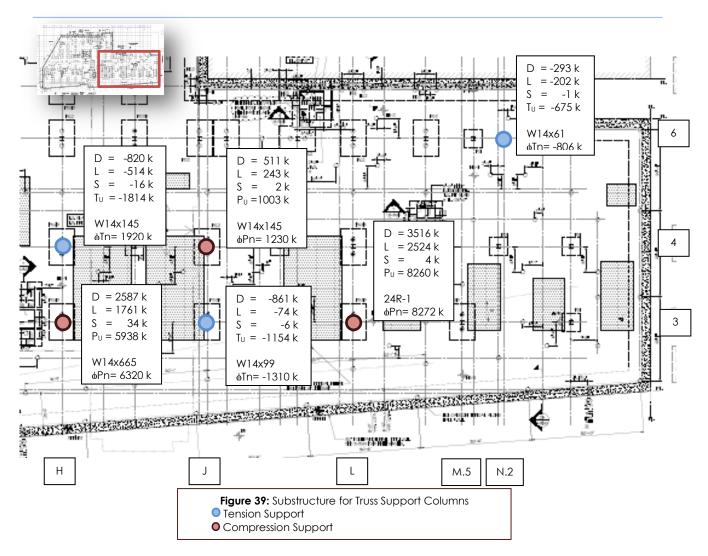
	ETA	BS	Hand	Error		
Frame	Shear	Axial	φPn	D	%D	
1	2005	3393	3077	315.8	-9.31	
2	2315	3724	3759	35.49	-0.95	
3	2400	3568	3690	122.2	+3.42	
4 1985 4635 4902 267 +5.76						
Figure 37: Diagonals ETABS/Hand Comparison						

IMPACT ON FOUNDATIONS

As the final stage in designing AAM's superstructure without Column 3-M.5, an analysis was performed to determine the adequacy of the current foundation design. First, the final support reactions were itemized from trusses H, J, and X, and columns were selected or designed to carry the required loads. Next, remaining loads from level 1 were added and the substructure established. Finally, the pile arrangements supporting each of the columns were re-evaluated to reflect the strength requirements of the proposed design.

Columns supporting the trusses were not considered as part of the truss system and were therefore not analyzed in ETABS. Applied loads, however reflect the ETABS reactions factored according to the load combination parameters described in the Load Path Overview section of this report. Figures 38 and 39 below show the factored loads and members selected for each of the affected columns in the superstructure and substructure respectively. Member sizes were selected based on a 25' un-braced length. More detailed calculations can be found in Appendices E (superstructure) and F (substructure).





The loads shown in Figure 39 were used to analyze the capacity of the current pile arrangements before a final proposal could be issued. The capacities of assigned pile identifications are shown in Figure 40 below.

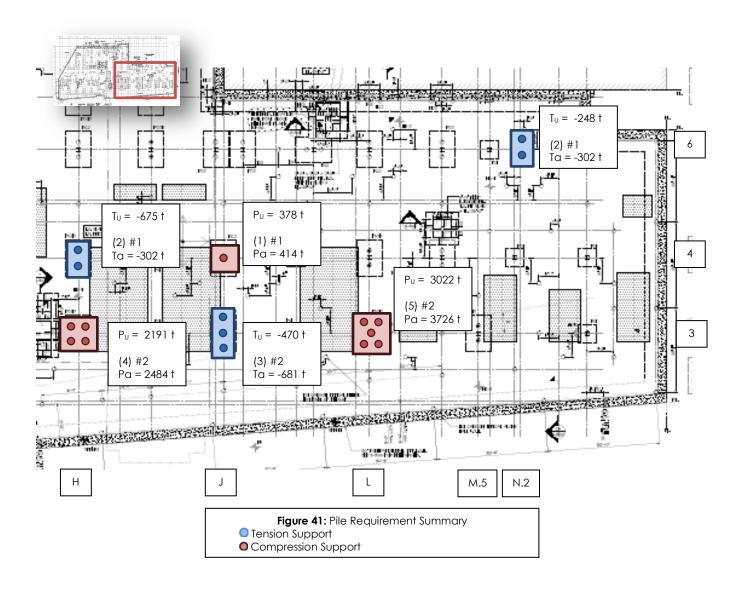
	CAISSON SCHEDULE									
	MARK	NOTES	CAISSON O.D.	CASING	CAISSON REINF. VERT. BARS	MIN. DEPTH OF ROCK SOCKET*	TENSION CAPACITY (TONS)	COMPRESSION CAPACITY (TONS)		
1	[o]	TYPICAL, @PC/WALL	13.375"	½" THICK Fy=80 ksi	I #24	11'-0"	151	4 4		
2	*	HIGH CAPACITY	13.375"	½" THICK Fy=80 ksi	2 #24	16'-0"	227	621		
3	0	TCI - NOT @PC/WALL	9.875"	½" THICK Fy=80 ksi	l #24	15'-0"	151	वा		

Figure 40: Caisson Schedule (FO-100)

An adequate number of piles were grouped to bear the loads from each column. While the columns were designed using LRFD ultimate loads shown in Figure 39 above, the foundation drawings do not contain notation that suggests LRFD was used. The pile capacities provided are therefore assumed to be based on ASD. Itemized column loads were simply added according to IBC Equation 16-9:

D+H+F+L+S+T

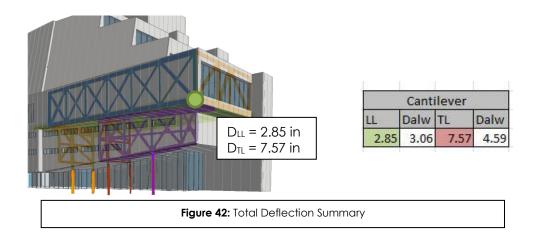
Figure 41 below summarizes the pile group requirements for the proposed structural system. This report does not include provisions for changing the capacity of the piles, but rather arranges the existing pile designs such that pile groups can adequately support the loads from above. Should the proposed system be accepted by the architect, the pile caps at the new pile groups will need to be designed as they were considered out of the scope of this investigation.



DEFLECTIONS AND SERVICEABILITY

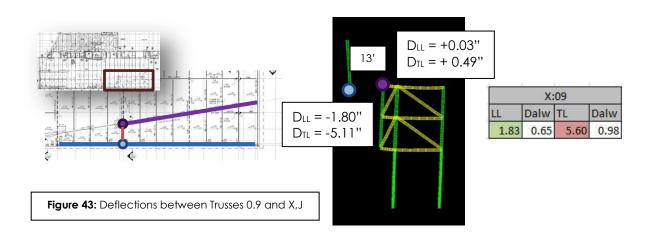
OVERALL DEFLECTION OF THE CANTILEVER

Though incremental deflections were not considered in the design of the trusses, the overall deflection at the cantilever was analyzed to determine the adequacy of AAM's proposed structural system. The allowable deflection at the cantilever was performed for both Live and Total load conditions using the shortest distance, 45'-10", to the last support at 3-L. Figure 42 below shows that deflections due to live loads were deemed acceptable, while the deflections due to total load fail by approximately 3". Further Calculations can be found in Appendix G.



SERVICEABILITY

In addition to the overall deflection of the cantilever, the trusses' close proximities also could create adverse effects on the serviceability of the structure. Figure 43 shows how the live load deflections were checked for proximity as well as span and cantilever length. At column line J level 5 experiences live load deflections in two different directions: up where Truss J supports Truss X and down where the floor is supported by Truss 0.9. The distance between these trusses is 13' (156") at this location, giving a maximum allowable LL deflection of 0.65" (I/360). In contrast to the cantilever, neither live load nort total load deflections between the highest point of Truss J and the closest deflected point at truss X pass. The deflections are so severe that the floor, wall, and ceiling materials risk damage. Furthermore, deflections of 6" over 13' would be visible under service dead and live loads.



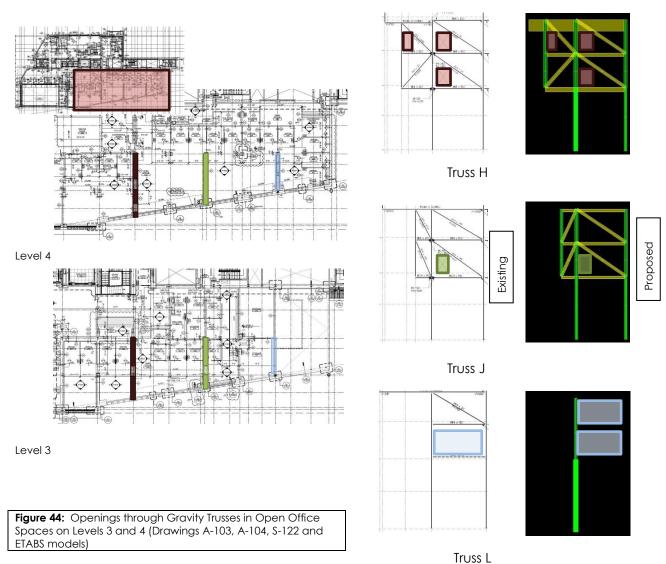
ARCHITECTURE CONSIDERATIONS

OVERVIEW

Respect for the current architectural scheme was a crucial consideration in the redesign of AAM. The office spaces on levels 3 and 4 are connected by passages through the existing truss systems. Entire systems and components such as section Truss N.2 and section PG56-1 were designed specifically to mitigate or eliminate clashes and alterations of the architecture. Some conflicts, such as Truss X's placement in front of office windows could not be avoided and will require further input from the architect.

LOWER TRUSSES: OPEN OFFICE SPACES

As mentioned above, the open office spaces on levels 3 and 4 are broken by gravity trusses which support the upper floors. In order to allow movement between these spaces, the web openings in the trusses were utilized by the architect. Figure 44 below displays how these openings were maintained in Trusses H and J for the proposed redesign. Additionally, Truss L was reduced to a single column, providing more flexibility for the open office space on level 4.



TRUSS N.2: LEVEL 5 GALLERY

PLACEMENT ALONG EAST WALL

Perhaps the most notable challenges with respect to the proposed structural system conforming to the architect's vision for AAM arise from the addition of truss N.2 (shown in Figure 45 right). Located between levels 5 and 6, proposed Truss N.2 marks the East end of the main gallery space of the museum. This main gallery was designed with a 200PSF live load, twice the code minimum for assembly spaces, and boasts 16,000 uninterrupted sq. ft. of space, made possible through a 70' span. All of these exceptional structural provisions were done to provide maximum flexibility for the space's use.



Figure 45: Location of Truss N.2

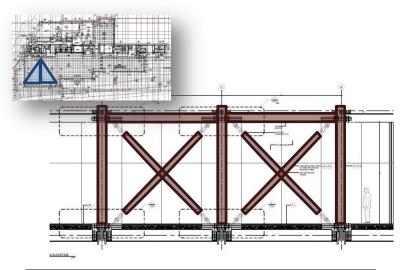
Another aspect of the uninterrupted space is the opportunity for long views which will provide relief to the public when visiting AAM. Large, uninterrupted windows were placed in the current design at the East and West walls of this main gallery. The East window will overlook the High Line park and city skyline, and the West will overlook the river and opposing shoreline (see Figure 2 in Building Introduction section). Proposed Truss N.2 is placed directly inside the East window, and would create a more obvious physical boundary between the gallery and its exterior view, while the West would appear to remain boundless to the river and beyond. Both the current and proposed designs can be seen in Figure 46 below.



Figure 46: Interior Renderings of Level 5 Gallery Space with Current (top) and Proposed (bottom) Designs (A-105)

ARCHITECTURAL USE OF STRUCTURE

While the addition of proposed Truss N.2 may conflict with the architectural aura of the main gallery space, there is precedence for exposing structural steel both in AAM and in Renzo Piano's other projects. Figure 47 provides an elevation of the exposed bracing in the level 1 gift shop, and Figure 48 shows the use of exposed structure in another Renzo Piano building.



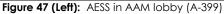


Figure 48 (Right): AESS used in Another Renzo Piano Building (courtesy of RPBW).



Renzo Piano's design for the II Sole 24 Ore headquarters in Milan, Italy (Figure 48) utilizes AESS in both the interior and exterior portions of the building. Furthermore, while AAM's main public space is the level 5 gallery, II Sole's most important space is its main lobby. Both buildings highlight their respective structures as vital to the architecture without being overbearing. This balance is achieved by using the slender, round sections, and by strictly adhering to the rhythmic architectural module.

Section and Module Inconsistencies

Truss N.2 could not be designed with the round sections described above for strength reasoned which are further discussed in the Proposed Structural Design section of the Final Report. Instead, Wide-Flange shapes were used to carry the large axial forces present within the truss.

As mentioned in the Building Introduction section above, the steel panels that dominate the façade of the building work on a 6'-8" module. AAM's entire exterior, as made evident by the East elevation shown in Figure 49 below, was composed for harmony between the glass panels and steel panels, conforming to the modular rhythm established by the architect.

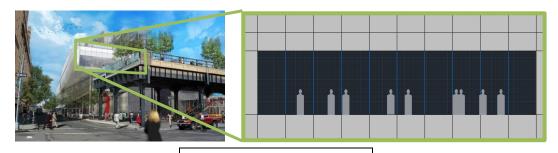
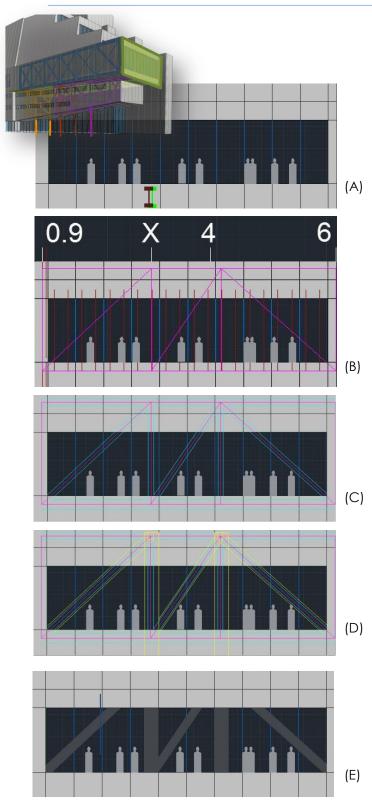


Figure 49: Current Façade Design



Proposed Truss N.2, however, cannot conform to the existing grid established by the façade and glazing panels. Figure 50-A (left) shows where Truss X intersects with and supports Truss N.2. Because column line N.2 lies slightly inward from the East wall, the intersection of the two column lines nearly coincides with the third-quarter point within the fourth wall panel (item B). Furthermore, the point of intersection creates an akward 26'-1" cantilever out to column line 0.9 from the support at Truss X.

In an effort to design a symmetric, yet rational truss for N.2, the other vertical member was placed at a more constructable 2'-6" north of column line 4 (see item B). This position was chosen because it very nearly coincides with the first-quarter point within the sixth wall panel.

While the two verticals are very close to a perfectly rational design that is consistent with the architecture, the small discrepancies of no more than 2" remain. The placement of the forward vertical member cannot be altered due to structural requirements, meaning that a symmetric and efficient design cannot be wholly reconciled to the panel system currently in place.

One option for creating the perfect alignments that are uniquely considered in Renzo Piano's architecture is an additional envelope around Truss N.2. A rational design of this envelope is shown in items D and E of Figure 50. The envelope first covers the columns on a halfpanel basis, keeping and perfecting the symmetry of the truss. Secondly, the diagonal envelopes extend from the corners of the rectangles that form from the intersection of the vertical members with the top chord of Truss N.2. This rational design reinforces the rhythm of the façade and minimizes the impact of the truss within the main gallery space. Furthermore, exposing Wide Flange sections within AAM would be inconsistent with the exposed HSS braces visible on level 1. Figure 51 below shows how the enveloped truss would appear inside the main gallery.

Figure 50: Truss N.2 Module Conflicts:

- (A) Intersection of Truss X and Truss N.2
- (B) Proposed Truss N.2

- (C) Interior Envelope Overlay
- (D) Interior Envelope Schematic
- (E) Proposed Alternative Truss Cover

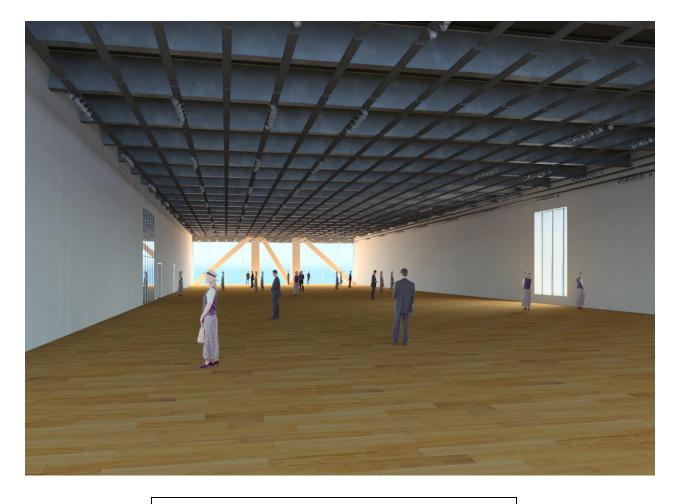


Figure 51: Enveloped Truss N.2 inside Level 5 Gallery

PG56-1: ARCHITECTURAL ENVELOPE

In order to accommodate the large structural members required in the cantilever, the architect allotted 5'-8" of space between the top of the floor on level 5 and the bottom of the architectural envelope as shown in Figure 53. When considering a 10"-thick floor system and a 2"-thick envelope structure, 4'-8" (56") remain as the absolute maximum thickness for a structural member.

The current design employs 46"-deep plate girders, which leaves an additional 10" of clearance for MEP systems. A 56"-deep member allows no additional space for MEP systems, meaning that web openings would need to be considered for actual use.

Furthermore, proposed section PG56-1 does not strictly adhere to the precedent plate components established in the current design. Both the plate thickness and width had to be increased to accommodate the loads. A section following the precedent plate sizes would need to be 76" deep to achieve the same strength as PG56-1 (see Appendix C). Adhering to the envelope limits established by the architect was the chief constraint in the design of PG56-1.

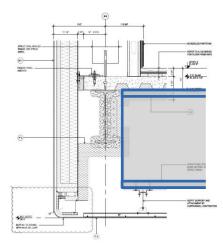




Figure 52: Section showing envelope dimensions at cantilever (3: A-357)



Plate Girder Outline

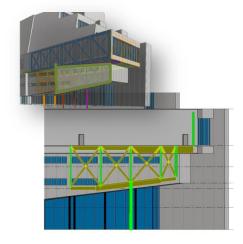
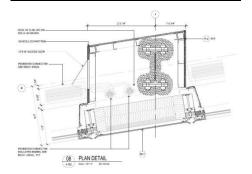


Figure 53 (Above):

Structural/architectural conflict on levels 3 and 4.

Figure 54 (Below): Exterior Wall Detail (A-352)



TRUSS X: LEVELS 3 AND 4 EXTERIOR WALL

As the third layer of the proposed load path and the primary support of the cantilever, Truss X carries the largest loads in AAM, thus requiring the X-bracing and two-story geometry shown. Though the horizontal spacing was held according to the current design, Figure 53 shows (left) shows how the diagonal braces could clash with the window placement in the open office spaces on levels 3 and 4.

Detail 08 on drawing A-352 (Figure 54 below) shows that the exterior face of the wall lies 24" outside of column line X, and the inside face is specified as against the fireproofing foam. The outside face will not be affected by Truss X; the widest shape, PG46-3, is 18" wide and will fit well within the exterior building envelope. Also, because the exterior face has no interference, the windows do not necessarily need to be moved or changed if a visible truss is deemed acceptable.

Since Figure 54 is based on a W14x column, the drywall will be pushed into the office space by 2" under the proposed structural system. Because the current design is so dependent on the windows it is difficult to judge how much square footage will be lost in these spaces.

CONSTRUCTION MANAGEMENT CONSIDERATIONS

OVERVIEW

Because the structural system proposed in this report was designed to minimize effects on the established architectural scheme, the construction of AAM will be both more expensive and more difficult. As briefly described in the Proposed Structural Design section above, the weight was increased and its distribution changed. Those alterations to the superstructure also affected the number and arrangement of the piles at the foundations. Additionally, the proposed structural system consists of long-span trusses (up to 122'), which will be difficult to both transport to the site and to lift into place. Finally, both proposed custom sections PG56-1 and 24R-1 will require special consideration for the procurement of elements and construction techniques.

Cost

SUPERSTRUCTURE

Cost data provided by a contact at Barton Malow Company assesses the cost of structural steel based on its overall weight, so a takeoff was performed to compare the weight of the current and proposed truss systems. The proposed structural system weighs nearly 100 t heavier than the current system. Where the heaviest element currently is Truss J at over 45 t, proposed Truss X weighs over 120 t alone. Figure 55 below summarizes the findings, and is broken down by congruent element.

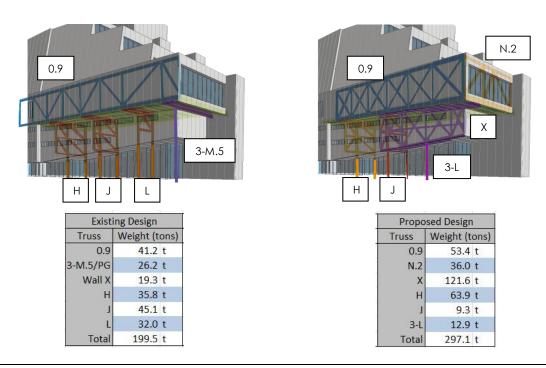


Figure 55: Weight Comparison of Existing (left) and Proposed (right)

Once the weight of both systems was established, the cost data was applied and an increase in cost of \$2,017,824 was found for the proposed system, which is summarized in Figure 56. The starred values were provided by Barton Malow from the company's 2012 cost database. Because this data is only for weight, and does not account for the location, timeframe, or specialty items, increases were added for both the current and proposed designs for a conservative estimate. Furthermore, location factors were taken from RS Means 2012 for the correct city (which the owner requested not to be disclosed). A time factor accounts for 1% inflation because the steel framing was built early in 2013, not in 2012. Finally an Overhead and Profit factor of 15% was added to determine the total cost of each system. More detailed calculations are provided in Appendix H.

	We	ight	ı	Material		-	abricating			Install				
System	tons	lbs	*0.80	cost	Loc	*2.50	cost	Loc	*2.75	cost	Loc	Time	0&P	Total Cost
Original	199.5	399045	0.80	319236	1.04	2.68	1069442	1.670	2.78	1109346	1.139	1.01	15%	3928375
Redesign	297.1	594224	0.80	475379	1.04	2.75	1634115	1.670	2.80	1663826	1.139	1.01	15%	5946200
												Total	\$	2017824

Figure 56: Superstructure Cost Comparison

FOUNDATIONS

Unlike the structural steel, no cost data was provided for the foundations, so the cost analysis for the piles was performed according to RS Means 2012. In order to use RS Means, however, the deep piles had to be taken off in terms of vertical linear feet. Figure 57 below shows a geologic section provided by the URS Geotechnical Investigation (2011). The end-bearing piles will rest on bedrock, which lies at an average depth of 90' for the site. Knowing that the bottom of the floor slab rests at a depth of 22', the piles must extend roughly 68' before being embedded into the bedrock.

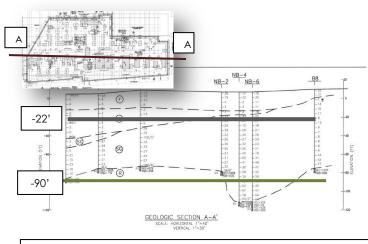


Figure 57: Geotechnical Section A-A

The Caisson Schedule on S-120.01 (see also Figure 40 in the Proposed Structural System section above) notes that each of the caisson types has a unique embedded length. Once a cost/linear foot value was established for the piles, the overall cost was determined by finding the total length of driven piles for each system. A summary is provided in Figure 58.

Comparing the number of each type of pile used determined the total cost for each foundation system because of the differing embedment lengths. The proposed foundation system costs nearly \$100,000 more than the current design.

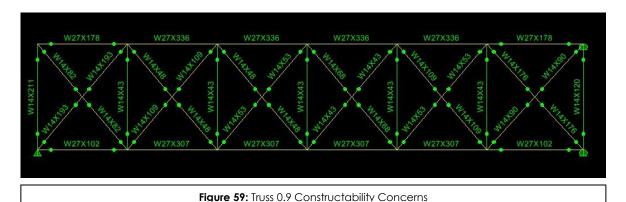
Current	Туре	n	Cost
	1	2	37722.01
	2	10	200547.39
	Total	\$	238269.40
Proposed	Туре	n	Cost
	1	5	94305.02
	2	12	240656.87
	Total	\$	334961.90
Diffe	rence	\$	96692.49

Figure 58: Foundations Cost Comparison

CONSTRUCTABILITY

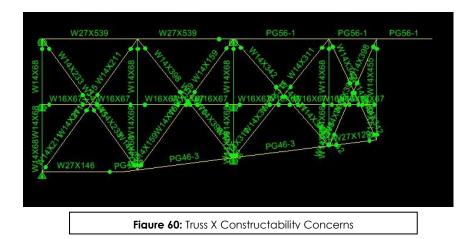
TRUSSES

TRUSS 0.9



Proposed Truss 0.9 spans 121.5' from gridline H to gridline N.2, is 23'-8" tall, and weighs over 53 t. It is highly unlikely, therefore, that a single crane could lift the whole truss into place. Furthermore, the city's access points, streets, and intersections are likely too low and too narrow to bring Truss 0.9 in by truck. In an effort to ease these constraints, and increase the structural efficiency of the truss, pin connections were added to separate the 4 interior panels from the 2 exterior panels (shown in Figure 59 above). This provision changes the longest span to 80', which may make truck transportation possible. If truck transportation remains impossible, however, the General Contractor will need to arrange for The Truss to be barged in on the river adjacent to the site (see Figure 2 in the Building Introduction section above).

Truss X



In a similar fashion to Truss 0.9, Truss X weighs over 120 t and spans nearly 85' from gridline J to gridline N.2. Instead of being broken horizontally, however, Truss X is designed with pin connections at Level 4, meaning the top and bottom halves of the truss could be brought in separately (Figure 60). Also, if truck transportation is not an option, then the truss will have to be brought in by barge.

CUSTOM MEMBERS

PG56-1

As described above in the Custom Cross Sections section of this report, Figure 61 shows how the dimensions of the plates used for PG56-1 depart from the precedent plate sizes established by the engineers (found on the Plate Girder Schedule). In order to accommodate the immense loads and avoid interfering with the architectural envelope, a base dimension, B, of 24" is proposed, 4" wider than the current largest width of 20" for PG46-2. Bending requirements led to a 10"-thick, built-up flange. This does not necessarily conflict with the established design philosophy, as PG46-2 specifies (2) 4"-thick plates be welded together. PG56-1 could simply be a modification of that flange by welding an additional 1"-thick plate, or another arrangement could be established. Finally, for shear purposes the maximum established web thickness of 2" was increased to 2 1/4". Because of these provisions, increases were made to the labor, installation, and fabrication costs of the structural steel for the cost estimate.

	PG56-:	1		Capaciti	ies
Lb	20	ft	φMn	41571	ft-k
D	56	in	φVn	3402	k
В	24	in	φTn	25245	k
tf	10	in	φPn	27541	k
tw	2.25	in			

BEAM		BEAM DIME	NSIONS	
MARK	В	tp	D	^t w
P632.5	18"	4"	32.5"	2"
P633-I	18"	4"	33"	2"
PG44-I	18"	4"	44"	2"
PG46-I	18"	2"	46"	Ţ n
PG46-2	20"	(2) 4" **	46"	2"
PG46-3	18"	4"	46"	2"
PG72-I	16"	3"	72"	2"

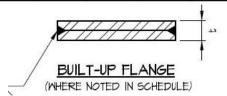


Figure 61: PG56-1 Precedence and Comparison (S-210)

24R-1

Unlike PG56-1, columns utilizing cross section 24R-1 face serious construction challenges. The section, detailed in Figure 62 below, specifies an unusually thick pipe wall of 1 ¾" with a 24" outer diameter. Attempts to contact the steel fabricator regarding this provision were unsuccessful.

The most risky specification for 24R-1, however, is the requirement for a concrete compressive strength of 15,000 psi. Though this reflects an extremely high compressive strength, it is not unprecedented in the United States. The Portland Cement Association's page High-Strength Concrete (see References below) notes that compressive strengths as high as 19,000 psi have been used in large cities like Seattle. The use of 15,000 psi concrete will also likely involve more testing and regulation, as the highest-strength concrete is currently specified at 5,000 psi at the foundations.

In addition to the difficulty acquiring and ensuring such a high compressive strength, the presence of reinforcement and containment in a steel pipe make workability an issue. Extra care will need to be taken by the general contractor and subcontractors to ensure the concrete is properly placed and vibrated to ensure the capacity of the columns.

	Pipe
Do	24 in
t	1.75 in
Co	oncrete
f'c	15000 psi
no.	11
n	16
C	apacity
φPn	8272 k
φTn	8053 k
φMn	2754 ft-k

Sean Felton | Structural Option | Advisor: Susteric | April 3, 2013

COMPARISON AND CONCLUSION

The proposed structural design contained in this report reflects a thorough investigation into the possibility of supporting the South-Eastern corner of AAM without the use of a column at 3-M.5. In order to both achieve structural adequacy under this constraint and minimize impacts to the architecture, the structural system becomes defined by its departure from common practice and precedent provisions. Figure 63 below shows that even a 50% increase in local weight, a 33% increase in cost, and radically high concrete strength specifications, AAM's proposed structural system fails in serviceability, unacceptably interferes with the window placement on levels 3 and 4, and causes serious logistical concerns during fabrication and construction. After assessing the impacts of the proposed structural system, it is recommended that AAM be constructed under the current design and specifications put forth by Robert Silman Associates.

Structural Concerns	Current Design	Proposed Design
Remove Column 3-M.5	NO	YES
No. of Steps in Load Path	2	4
Max. Element Weight	45.1 t	121.6 t
Overall Weight	199.5 t	297.1 t
Max. Pile Group	2	5
No. of Custom Sections	10	12
Columns Max.O.D.	22"	24"
Max. f'c	5,000	15,000
Max. Total Deflection	126	-7.57 in
Acceptable Deflections	YES	NO
Architectural Concerns	Current Design	Proposed Design
Gallery Interference	NO	Truss N.2
Wall X Interference	NO	Truss X
Remove Truss L	NO	YES
Maintain Web Openings	YES	YES
Maintain Building Envelope	YES	YES
Construction Concerns	Current Design	Proposed Design
No. of Long Trusses	1	2
Cost of Superstructure	\$3,928,000	\$5,946,000
Cost of Foundations	\$238,000	335,000
Total Structural Cost	\$4,166,000	\$6,281,000
Total Difference		\$2,115,000

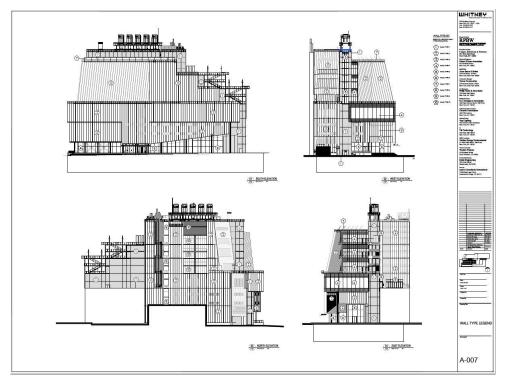
Figure 63: Comparative Summary

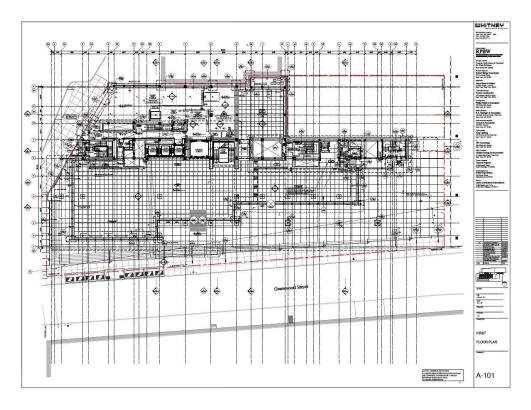
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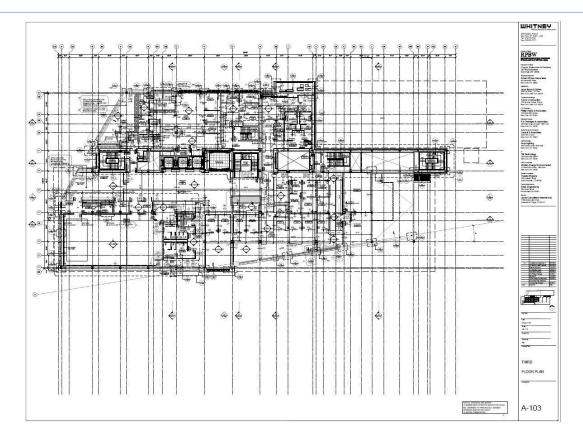
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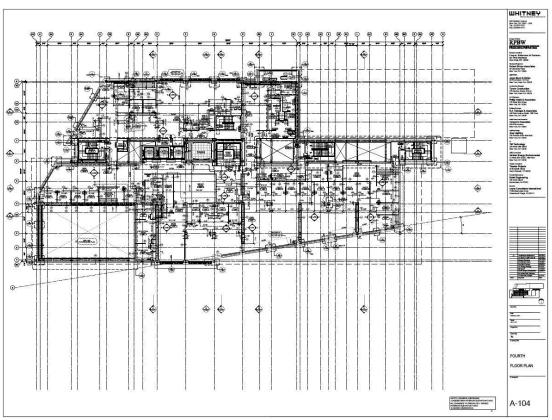
APPENDIX A: TECHNICAL DRAWINGS

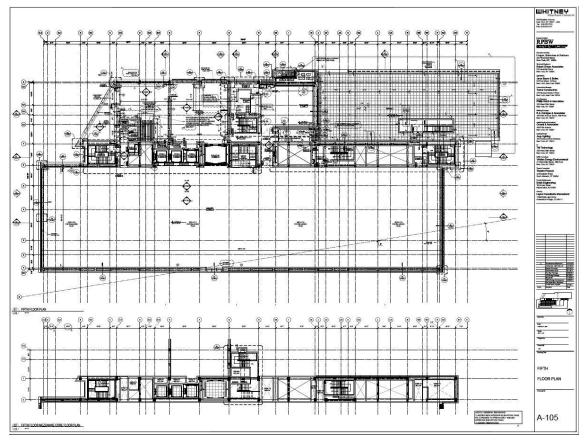
ARCHITECTURAL DRAWINGS

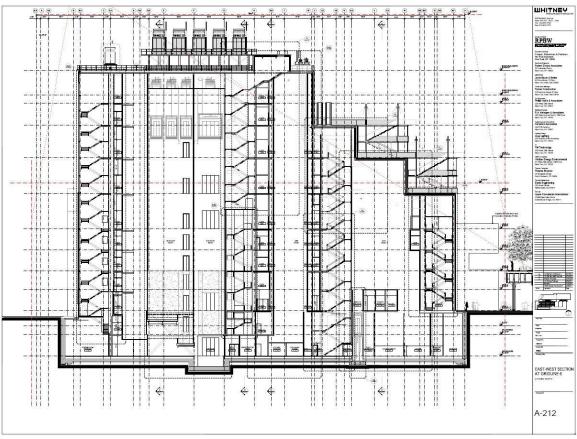




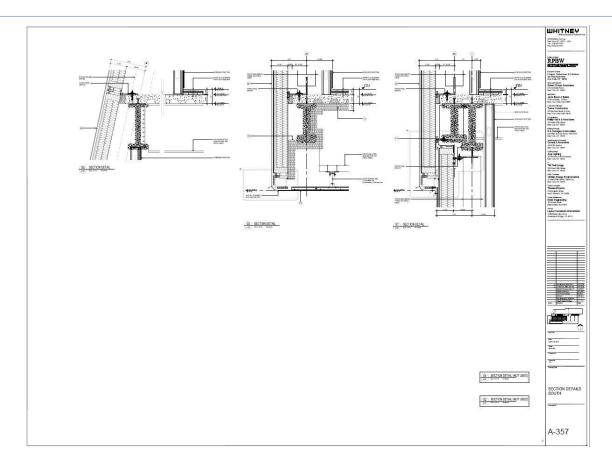


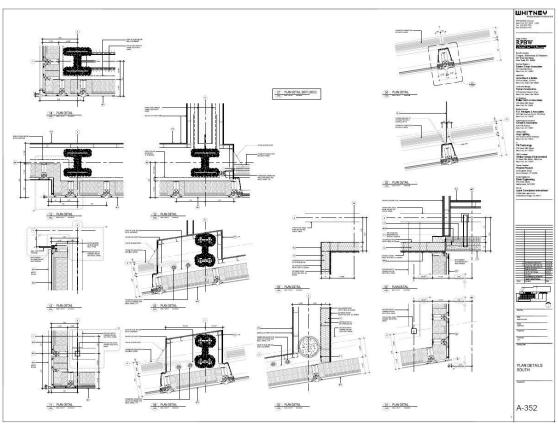




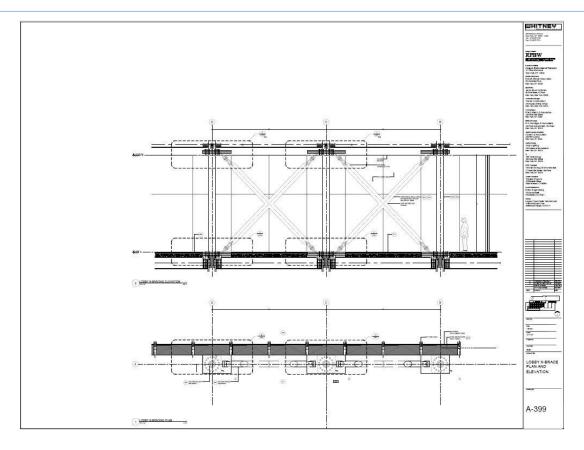


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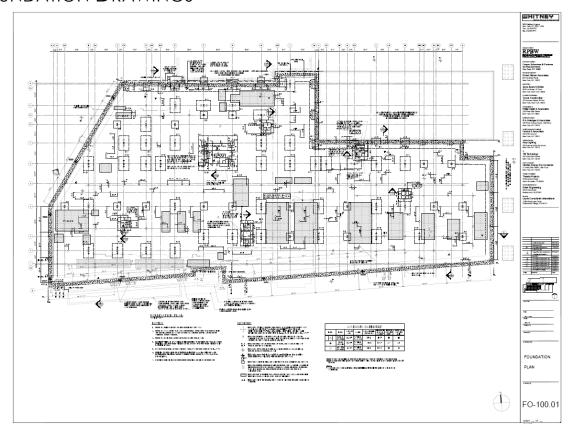




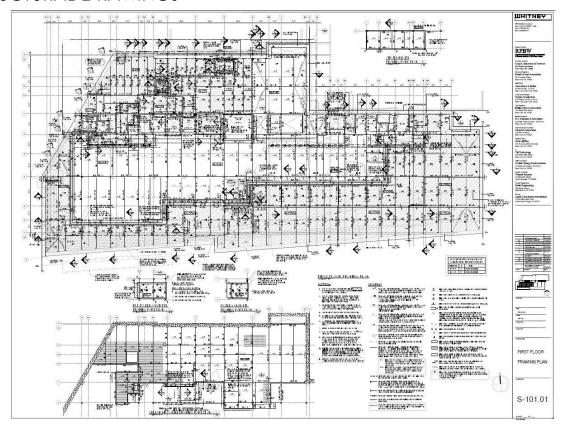
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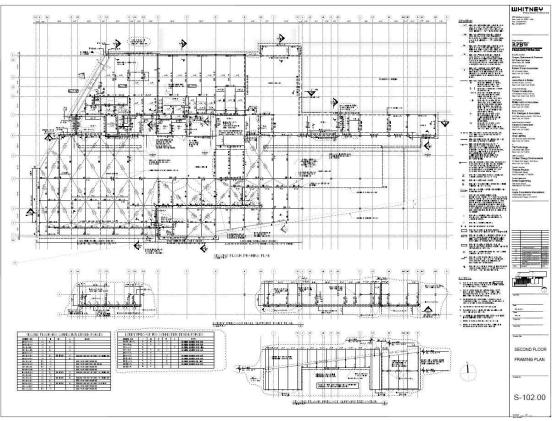


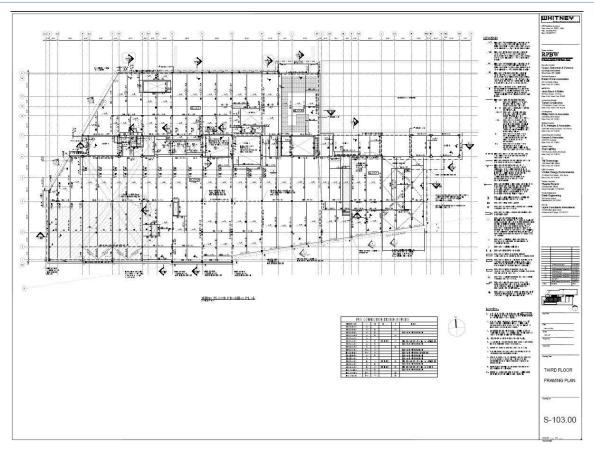
FOUNDATION DRAWINGS

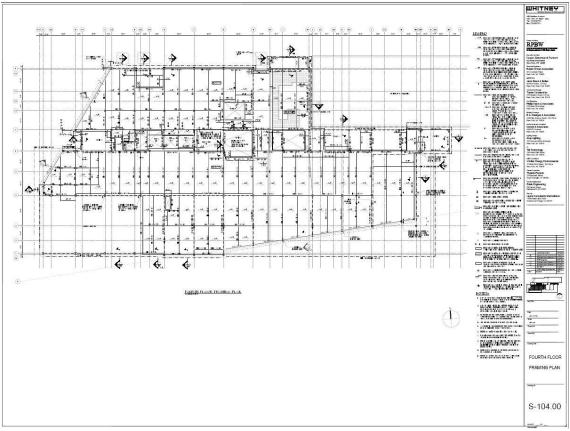


STRUCTURAL DRAWINGS

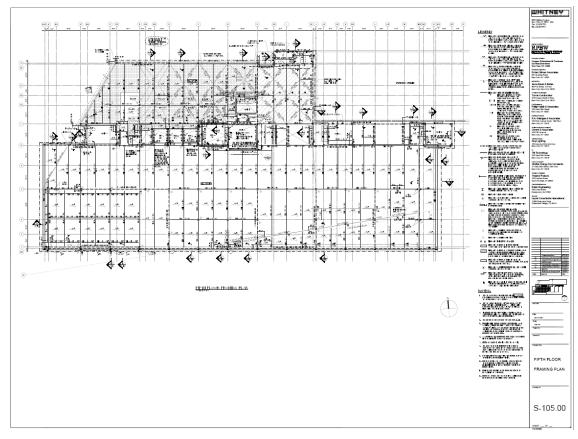


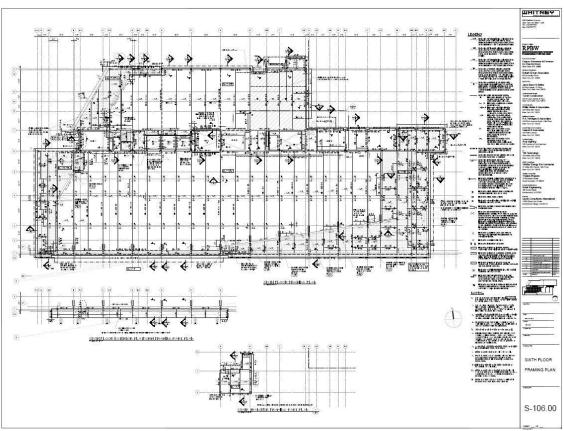


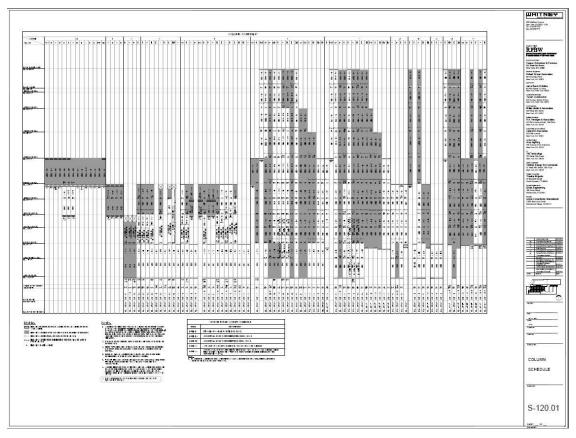


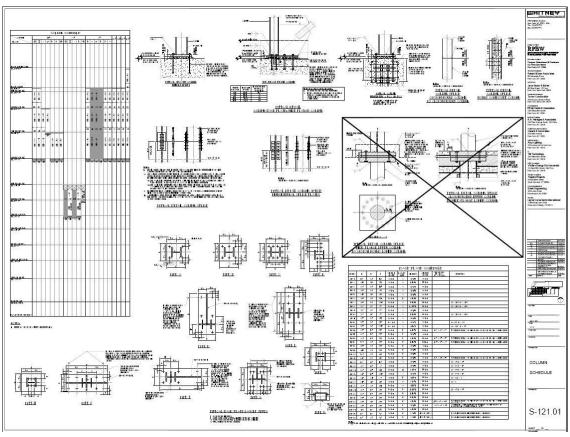


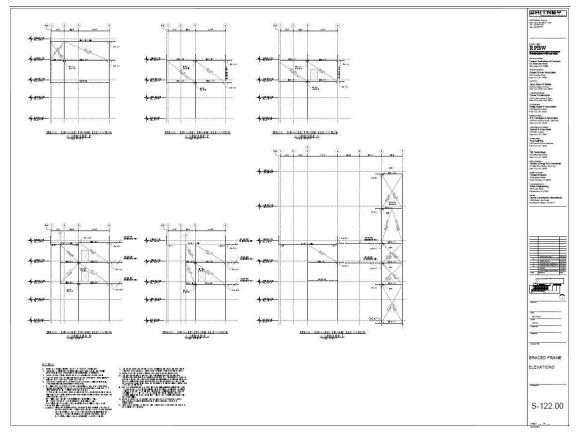
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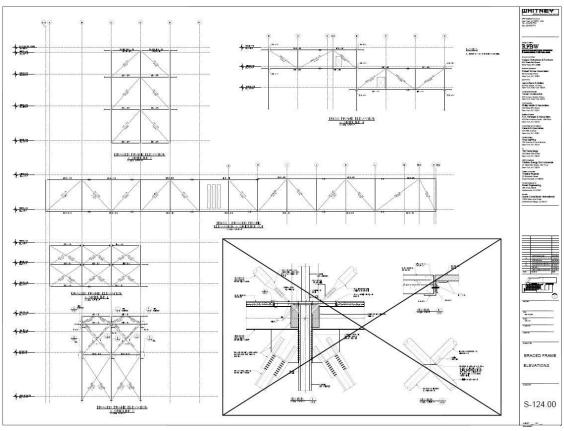


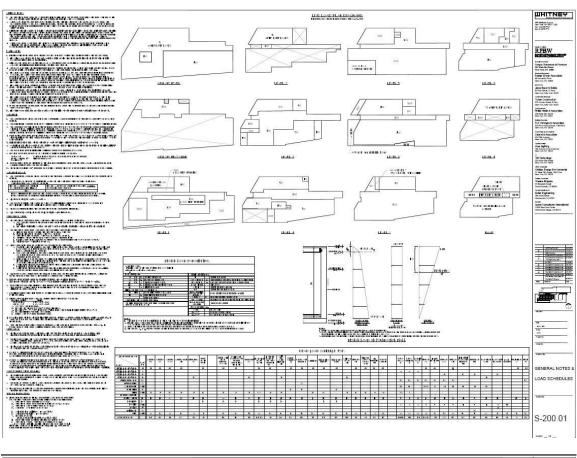


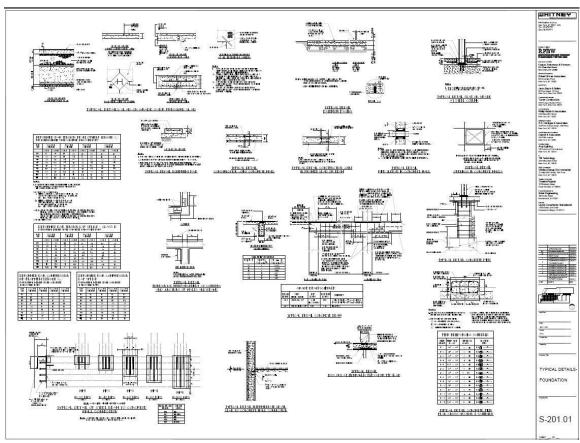




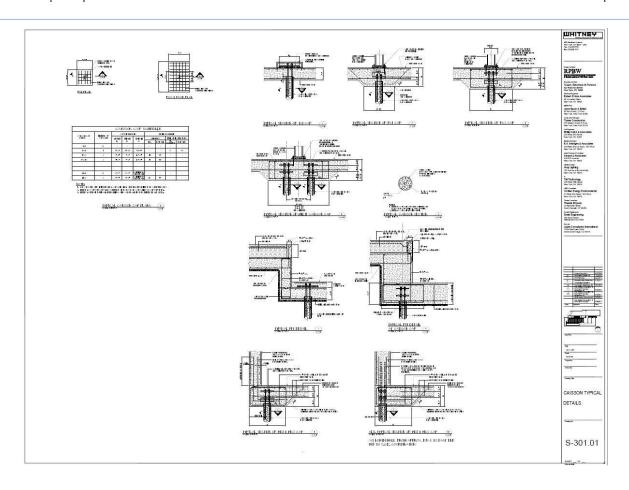




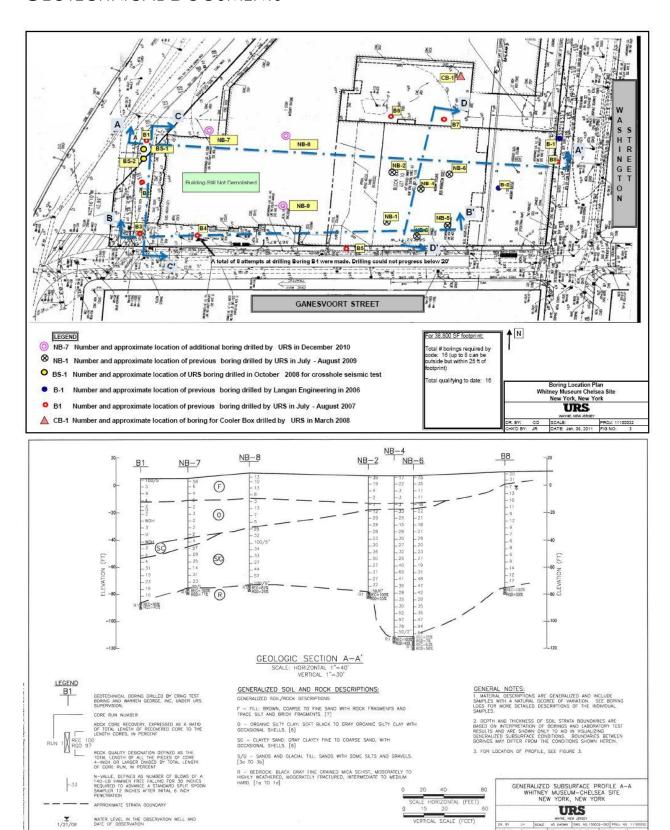




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GEOTECHNICAL DOCUMENTS



APPENDIX B: INTRODUCTION CALCULATIONS

BUILDING DEAD LOAD CALCULATIONS

		То	tal Dead Load	Calculation	ons	
Level	Туре		SQ in	SQ ft	Wt/SFt	Wt/flr (k)
Roof N	31		431080	2994	102	305.35
483	32		220480	1531	116	177.61
Roof S	33		154530	1073	161	172.77
628	34		128723	894	118	105.48
	35		598722	4158	84	349.25
Level 9	16		96701	672	99	66.48
500	37		495578	3442	126	433.63
Level 8	3		877728	6095	121	737.54
2002	6		119746	832	98	81.49
	7		225656	1567	118	184.91
	8		271800	1888	116	218.95
	16		415454	2885	99	285.62
	23		75238	522	112	58.52
	27		334730	2325	187	434.68
Level 7	3		1498650	10407	121	1259.28
2342	6		535436	3718	98	364.39
	8		69584	483	116	56.05
	12		123266	856	98	83.89
	16		40078	278	99	27.55
	20		83450	580	94	54.47
	21		103600	719	84	60.43
	27		335340	2329	187	435.48
Level 6	2		1897600	13178	136	1792.18
4188	4		460080	3195	107	341.87
	12		49612	345	98	33.76
	13		79600	553	166	91.76
	16		40078	278	99	27.55
	19		103640	720	154	110.84
	28		156520	1087	214	232.61
	29		988974	6868	203	1394.18
	30		149084	1035	158	163.58

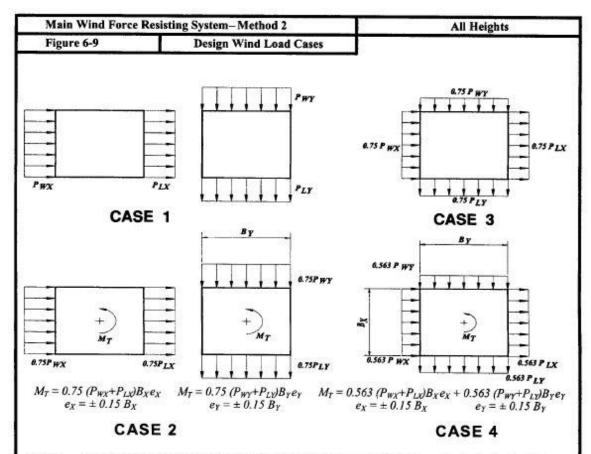
	1	Total Dead Loa	ad Calculati	ons	
Level	Туре	SQ in	SQ ft	Wt/SFt	Wt/flr (k)
Level 5	1	2830400	19656	109	2142.46
2915	5	172000	1194	158	188.72
	11	84200	585	133	77.77
	16	40078	278	99	27.55
	22	564400	3919	122	478.17
Level 4	6	2801124	19452	98	1906.32
2589	8	90400	628	116	72.82
	10	93800	651	109	71.00
	12	98340	683	98	66.93
	16	591510	4108	99	406.66
	23	84280	585	112	65.55
Level 3	6	949600	6594	98	646.26
2155	7	93200	647	118	76.37
	8	205328	1426	116	165.40
	9	458320	3183	181	576.08
	12	26000	181	98	17.69
	16	704038	4889	99	484.03
	23	243288	1690	112	189.22
Level 2	16	265600	1844	99	182.60
419	36	448300	3113	76	236.60
Level 1	14	1434000	9958	126	1254.75
4863	15	371600	2581	148	381.92
	16	222000	1542	99	152.63
	24	1222800	8492	186	1579.45
	25	384200	2668	191	509.60
	26	839400	5829	169	985.13
				Totals	
			Sq. ft		Weight (k)
			183882		23084

DL Sche	dule Sun	nmary (S-200.0	01)
	DL		DL
Floor Type	PSF	Floor Type	PSF
1	109	21	84
2	136	22	122
3	121	23	112
4	107	24	186
5	158	25	191
6	98	26	169
7	118	27	187
8	116	28	214
9	181	29	203
10	109	30	158
11	133	31	102
12	98	32	116
13	166	33	161
14	126	34	118
15	148	35	84
16	99	36	76
17	124	37	126
18	135		
19	154		
20	94		

SNOW LOAD CALCULATIONS

ASCE 7-05 SNOW LOADS	1/1
GROUND SNOW LOAD: 7.2	
PO = 25 PSP (FIGURE 7-1) - verified by smutural notes, city ac	
FLAT ROOF SNOW LOADS 7.3	
P4 = 0.7 Ce Ct I pg , P+min = { I pg (pg & 20 PSF) }	
EQN 7-1	
EXPOSURE FACTOR 7.8.1	
TERRAIN CATEGORY "L" (WIND CALLS PI, AGLETION 6.5.6) PARTIALY EXPOSED ROOF	
(e = 1.0 (TABLE 7-2) - werifted by smutural notes	
THERMAL FALTOR 7.3.2	
(t=1.0 LTABLE 7-3) - verified by smithal notes	
IMPORTANCE FACTOR 7.3.3 OCCUPANCY III T = 1.15 - verifica lag ancieval notes	
P4 = 0.7 . 1.0 1.0 . 1.15 . 25 = 21 PSF	
Pf. vin = 20 1 = 20.115 = 23 PSF * CONTROLS	
* DRAWINGS DENOTE 22 PSF FOR P4. LIKELY ASSUME LL	

WIND LOAD CALCULATIONS



- Case 1. Full design wind pressure acting on the projected area perpendicular to each principal axis of the structure, considered separately along each principal axis.
- Case 2. Three quarters of the design wind pressure acting on the projected area perpendicular to each principal axis of the structure in conjunction with a torsional moment as shown, considered separately for each principal axis.
- Case 3. Wind loading as defined in Case 1, but considered to act simultaneously at 75% of the specified value.
- Case 4. Wind loading as defined in Case 2, but considered to act simultaneously at 75% of the specified value.

Notes:

- Design wind pressures for windward and leeward faces shall be determined in accordance with the provisions of 6.5.12.2.1 and 6.5.12.2.3 as applicable for building of all heights.
- 2. Diagrams show plan views of building.
- 3. Notation:

 P_{WX} , P_{WY} : Windward face design pressure acting in the x, y principal axis, respectively.

 P_{LX} , P_{LY} : Leeward face design pressure acting in the x, y principal axis, respectively.

 $e(e_X, e_Y)$: Eccentricity for the x, y principal axis of the structure, respectively.

M_T: Torsional moment per unit height acting about a vertical axis of the building.

		Wa	II Pressur	es				1	Eq	uivalent	Point Load	ds	
			qGfCp	qiGCpi	ww	LW	Pressure	E-W	Pressure	hi	Dist Ld	Bx	Px
ht	Kz	qz	Ср	-0.55	0.8	-0.3	PSF	Level	PSF	ft	plf	ft	k
160	1.39	33.41	23.793	-18.37	42.17	-7.72	49.89	RN	49.89	10	498.9	55.8	27.
142	1.36	32.68	23.279	-17.98	41.26	-7.72	48.98	RS	48.98	9	1348.3	53.8	72.
140	1.36	32.68	23.279	-17.98	41.26	-7.72	48.98	9	48.98	18	1789.1	55.8	99.
124	1.32	31.72	22.595	-17.45	40.04	-7.72	47.76	8	47.76	19	1964.3	113.7	223
102	1.26	30.28	21.568	-16.65	38.22	-7.72	45.94	7	45.94	23	1589.9	118.3	188.
78	1.21	29.08	20.712	-15.99	36.71	-7.72	44.43	Vb =	611.6	(Mover =	27902.3	kft
	10	Wa	II Pressur	es					Eq	uivalent	Point Load	İs	
100	90		qGfCp	qiGCpi	ww	LW	Pressure	N-S	Pressure	ht	Dist Ld	Ву	Ру
ht	Kz	qz	Ср	-0.55	0.8	-0.5	PSF	Level	PSF	ft	plf	ft	k
160	1.39	33.41	22.631	-18.37	41.00	-13.38	54.38	RN	54.38	10	543.8	150.3	81.
142	1.36	32.68	22.143	-17.98	40.12	-13.38	53.50	RS	53.50	9	1527.9	143.8	219.
140	1.36	32.68	22.143	-17.98	40.12	-13.38	53.50	9	53.50	18	2009.4	150.3	301.
124	1.32	31.72	21.491	-17.45	38.94	-13.38	52.32	8	52.32	20	2209.0	191.2	422.
102	1.26	30.28	20.515	-16.65	37.17	-13.38	50.55	7	50.55	23	1162.6	229.2	266.
	1.21	29.08	19.700	-15.99	35.69	-13.38	49.07	Vb =	1292.1		Mover =	65303.3	kft
	160 142 140 124 102 78 ht 160 142 140	160 1.39 142 1.36 140 1.36 124 1.32 102 1.26 78 1.21 ht Kz 160 1.39 142 1.36 140 1.36	ht Kz qz 160 1.39 33.41 142 1.36 32.68 140 1.36 32.68 124 1.32 31.72 102 1.26 30.28 78 1.21 29.08 Wa ht Kz qz 160 1.39 33.41 142 1.36 32.68 140 1.36 32.68	QGfCp ht Kz qz Cp 160 1.39 33.41 23.793 142 1.36 32.68 23.279 140 1.36 32.68 23.279 124 1.32 31.72 22.595 102 1.26 30.28 21.568 78 1.21 29.08 20.712	Ht Kz qz Cp -0.55	QGFCP QIGCPi WW	QGfCp QiGCpi WW LW	Name	No. No.	The image	Red Red	QGfCp QiGCpi WW LW Pressure E-W Pressure hi Dist Ld	Ht Kz qz Cp -0.55 0.8 -0.3 PSF Level PSF ft plf ft

SEISMIC LOAD CALCULATIONS

			Se	Seismic Loads	2	30.			Ė	E-W Direction	ion		0 8		N-S Direction	tion	1940
	Ht (ft)	hi	W (k)	wh ^k	CVX	ĮĮ.	ΙΛ	Bx	5%Bx	Ax	RS	Mz (ft-k)	Ву	5%By	Ay	RS	Mzy (ft-k)
RN	160	20	841	841 21539881	0.2278	192	192	150	7.5	22.7		32649	99	2.8	7.1		3809
RS	142	18	649	13092727	0.1385	90	282	144	7.2	70.2	13.2	8558	54	2.7	278.9	1.0	242
6	140	16	829	13281336	0.1405	95	377	150	7.5	11.6		8326	99	2.8	11.4		3040
8	124	22	1674	1674 25746066	0.2723	456	833	191	9.6	2.3		9942	114	5.7	2.8	1 6	7333
7	102	24		2007 20876383	0.2208	443	1276	229	11.5	1.0		5077	118	5.9	1.0		2622
k	2	Σ	5849	5849 94536394		= qA	1276										
-	1.53	S				Mov =	158514										

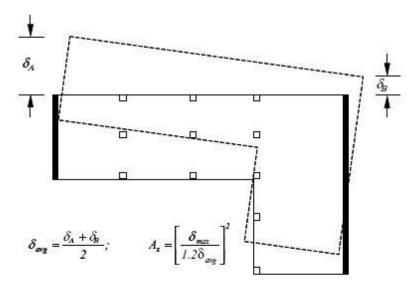


FIGURE 12.8-1 TORSIONAL AMPLIFICATION FACTOR, A.

353		Amplific	ation Facto	r Analysis		2	20.			RS Alter	native		
E-W	$(\delta_{xe})_1$	Δ_{xi}	(δ _{xe}) ₂	Δ_{x2}	Δ_{avg}	$\Delta_{\rm max}/\Delta_{\rm avg}$		$(\delta_{xe})_1$	Δ_{xi}	(δ _{xe}) ₂	Δ_{x2}	Δ_{avg}	$\Delta_{\sf max}/\Delta_{\sf avg}$
RN	1.057	0.236	1.054	0.134	0.1850	1.28	*						
RS	0.821	0.041	0.920	0.142	0.0915	1.55	*	0.556	0.128	0.705	0.195	0.1615	1.21
9	0.780	0.191	0.778	0.19	0.1905	1.00							
8	0.589	0.32	0.588	0.33	0.3250	1.02		0.428		0.510			
7	0.269	0.269	0.258	0.258	0.2635	1.02			-			-	
	$(\delta_{ye})_1$	$\Delta_{ t yi}$	$(\delta_{ye})_2$	Δ_{y2}	Δ_{avg}	$\Delta_{\sf max}/\Delta_{\sf avg}$		$(\delta_{ye})_1$	Δ_{yi}	$(\delta_{ye})_2$	Δ_{y2}	Δ_{avg}	$\Delta_{\sf max}/\Delta_{\sf avg}$
RN	0.183	0.053	-0.238	-0.102	-0.0245	4.16	*			1			
RS	0.130	0.01	-0.136	-0.016	-0.0030	5.33	*	0.308	0.135	0.274	0.100	0.1175	1.13
9	0.120	0.041	-0.120	-0.061	-0.0100	6.10	*						
8	0.079	0.042	-0.059	-0.043	-0.0005	86.00	*	0.173		0.174			
7	0.037	0.037	-0.016	-0.016	0.0105	3.52	*						
N-S	$(\delta_{xe})_1$	Δ_{x1}	$(\delta_{xe})_2$	Δ_{x2}	Δ_{avg}	$\Delta_{\rm max}/\Delta_{\rm avg}$	100	$(\delta_{xe})_1$	Δ_{x1}	(δ _{xe}) ₂	Δ_{x2}	Δ_{avg}	$\Delta_{\sf max}/\Delta_{\sf avg}$
RN	0.185	0.032	0.183	0.032	0.0320	1.00			8	2			
RS	0.153	0.007	0.151	0.006	0.0065	1.08		0.062	0.023	-0.045	-0.004	0.0095	2.42
9	0.146	0.03	0.145	0.03	0.0300	1.00							
8	0.116	0.038	0.115	0.037	0.0375	1.01		0.039		-0.041			
7	0.078	0.078	0.078	0.078	0.0780	1.00							
	$(\delta_{ye})_1$	$\Delta_{ t yi}$	(δ _{ye}) ₂	Δ_{y2}	Δ_{avg}	$\Delta_{\sf max}/\Delta_{\sf avg}$		$(\delta_{ye})_1$	Δ_{yi}	$(\delta_{ye})_2$	Δ_{y2}	Δ_{avg}	$\Delta_{\rm max}/\Delta_{\rm avg}$
RN	0.123	-0.829	0.978	0.218	-0.3055	2.71	*						
RS	0.952	0.053	0.760	0.042	0.0475	1.12		1.039	0.284	1.087	0.341	0.3125	1.09
9	0.899	0.245	0.718	0.198	0.2215	1.11							
8	0.654	0.342	0.520	0.306	0.3240	1.06	4	0.755		0.746			
7	0.312	0.312	0.214	0.214	0.2630	1.19							

	1	9	Seismic Desi	gn Criteria		
		S-20	0.01	ASCE 7	7-05	
		Sds	0.65	Ta (s)	0.9	
		Sd1	0.13	Cu	1.7	
		Ti Ti	1.25	T (s)	1.53	
		R	3	TL(s)	6	
		W (k)	5849			
		Cs	0.0602			
А	mp Facto	or Maximu	ıms	RS	Alternativ	/A
	δ _{EW}	Δ_{avg}	$\Delta_{\sf max}/\Delta_{\sf avg}$	δ _{EW}		Δ_{max}/Δ_{av}
RN		-	and the same of th	1	Δ_{avg}	
20000	δ_{EW}	Δ_{avg}	$\Delta_{\rm max}/\Delta_{\rm avg}$	1		$\Delta_{\sf max}/\Delta_{\sf av}$
RN	δ _{EW}	Δ _{avg} 0.185	$\Delta_{\sf max}/\Delta_{\sf avg}$ 5.71	δ_{EW}	Δ_{avg}	$\Delta_{\sf max}/\Delta_{\sf av}$
RN RS	δ _{EW} 1.057 0.920	Δ _{avg} 0.185 0.092	$\Delta_{\rm max}/\Delta_{\rm avg} = 5.71 = 10.05$	δ_{EW}	Δ_{avg}	$\Delta_{\sf max}/\Delta_{\sf av}$
RN RS 9	δ _{EW} 1.057 0.920 0.780	Δ _{avg} 0.185 0.092 0.191	$\Delta_{\rm max}/\Delta_{\rm avg}$ 5.71 10.05 4.09	δ_{EW}	Δ_{avg}	$\Delta_{\sf max}/\Delta_{\sf av}$
RN RS 9	δ _{EW} 1.057 0.920 0.780 0.589	Δ _{avg} 0.185 0.092 0.191 0.325	Δ _{max} /Δ _{avg} 5.71 10.05 4.09 1.81	δ_{EW}	Δ_{avg}	Δ_{max}/Δ_{av}
RN RS 9	δ _{EW} 1.057 0.920 0.780 0.589 0.269	Δ _{avg} 0.185 0.092 0.191 0.325 0.264	$\Delta_{\text{max}}/\Delta_{\text{avg}}$ 5.71 10.05 4.09 1.81 1.02	δ _{EW}	Δ _{avg} 0.1615	Δ _{max} /Δ _{av}
RN RS 9 8	δ _{EW} 1.057 0.920 0.780 0.589 0.269 δ _{NS}	$\begin{array}{c} \Delta_{\rm avg} \\ 0.185 \\ 0.092 \\ 0.191 \\ 0.325 \\ 0.264 \\ \Delta_{\rm avg} \end{array}$	$\Delta_{\rm max}/\Delta_{\rm avg}$ 5.71 10.05 4.09 1.81 1.02 $\Delta_{\rm max}/\Delta_{\rm avg}$	δ _{EW}	Δ _{avg} 0.1615	Δ_{max}/Δ_{av} 1.2 Δ_{max}/Δ_{av}
RN RS 9 8 7	δ _{EW} 1.057 0.920 0.780 0.589 0.269 δ _{NS} 0.978	$\Delta_{\rm avg}$ 0.185 0.092 0.191 0.325 0.264 $\Delta_{\rm avg}$ -0.31	$\Delta_{\rm max}/\Delta_{\rm avg}$ 5.71 10.05 4.09 1.81 1.02 $\Delta_{\rm max}/\Delta_{\rm avg}$ 3.20	δ _{EW} 0.705	Δ_{avg} 0.1615 Δ_{avg}	Δ_{max}/Δ_{av} 1.2 Δ_{max}/Δ_{av}
RN RS 9 8 7 RN RS	δ _{EW} 1.057 0.920 0.780 0.589 0.269 δ _{NS} 0.978 0.952	$\Delta_{\rm avg}$ 0.185 0.092 0.191 0.325 0.264 $\Delta_{\rm avg}$ -0.31 0.05	$\Delta_{\rm max}/\Delta_{\rm avg}$ 5.71 10.05 4.09 1.81 1.02 $\Delta_{\rm max}/\Delta_{\rm avg}$ 3.20 20.04	δ _{EW} 0.705	Δ_{avg} 0.1615 Δ_{avg}	

APPENDIX C: CUSTOM MEMBER STRENGTHS

MEMBERS SPECIFIED IN CURRENT DESIGN

PLATE GIRDERS

-	Plate		IL DITTE	ensions (i	117	(c
Shape	В	tf	nf	D	tw	h
32.5	18	4	1	32.5	2	24.5
33-1	18	4	1	33	2	25
44-1	18	4	1	44	2	36
46-1	18	2	1	46	1	42
46-2	20	4	2	46	2	30
46-3	18	4	1	46	2	38
72-1	16	3	1	72	2	66
L	20	ft				
Weld size	0.5	in				

Mater	rial Propo	erties
ф	0.9	
E	29000	ksi
Fy	50	ksi
G	11200	ksi
K	1.0	

Note: Dimensions taken from drawing S-211

							S	ection Pro	perties							
				Individu	ial Flan	ge			We	b	Total					
	Α	Х	Ах	Ax2	у	Ay2	lox	loy	А	Ax	А	Ixx	lyy	rx	ry	WT (plf)
32.5	72	14.25	1026	14620.5	5	1458.0	96.0	1944.0	49	600	193	29433	6804	12	6	656
33-1	72	14.5	1044	15138.0	5	1458.0	96.0	1944.0	50	625	194	30468	6804	13	6	659
44-1	72	20	1440	28800.0	.5	1458.0	96.0	1944.0	72	1296	216	57792	6804	16	6	734
46-1	36	22	792	17424.0	5	729.0	12.0	972.0	42	882	114	34872	3402	17	5	388
46-2	160	19	3040	57760.0	5	4000.0	853.3	5333.3	60	900	380	117227	18667	18	7	1291
46-3	72	21	1512	31752.0	- 5	1458.0	96.0	1944.0	76	1444	220	63696	6804	17	6	748
72-1	48	34.5	1656	57132.0	4	768.0	36.0	1024.0	132	4356	228	114336	3584	22	4	775

Tensile Strength									
	φTn (k)								
32.5	8685								
33-1	8730								
44-1	9720								
46-1	5130								
46-2	17100								
46-3	9900								
72-1	10260								

					Fle	exural St	rength						
	Flang	ge Compa	actness		Web	Compac	tness		Yie	lding	Max Unbr Length (Lp		
	bf/2tf	λp	λr		h/tw	λр	λr		Z	φMn (ft-k)	in	ft	
32.5	2.25	9.2	24.1	С	12.3	90.6	137.3	С	3252.5	12197	214	18	
33-1	2.25	9.2	24.1	С	12.5	90.6	137.3	С	3338	12518	214	18	
44-1	2.25	9.2	24.1	С	18.0	90.6	137.3	С	5472	20520	202	17	
46-1	4.50	9.2	24.1	С	42.0	90.6	137.3	С	3348	12555	197	16	
46-2	2.50	9.2	24.1	С	15.0	90.6	137.3	С	7880	29550	253	21	
46-3	2.25	9.2	24.1	С	19.0	90.6	137.3	С	5912	22170	201	17	
72-1	2.67	9.2	24.1	С	33.0	90.6	137.3	С	12024	45090	143	12	

30			=30	8 3				8 -	S 5	
		strength	∳Pn	8154 k	8193 k	9053 k	60	16445 k	9211 k	9291 k
		Compressive Strength	Pn	0906	9103	10059	28	18273	10235	10323
		Comp	Fcr	46.9	46.9	46.6	20	48.1	46.5	45.3
			Fcr tors	46.9	46.9	46.6	90.	48.1	46.5	45.3
		ng	Fe tors Fo	332	329	294	C	536	290	211
		Torsional Buckling	J F	460	461	491	20	3515	496	328
		Torsion	Cw	1381637	1430541	40 2721600	99	38 6738667	3000564	69 4265856
		- 0	ho	29	29	40	90	38	42	69
			Fcr flex	48.6	48.7	49.2	53	49.3	49.3	49.6
	h	kling	Fe flex	758	780	1330	C.	1533	1439	2492
	Compressive Strenth	Flexural Buckling	KL/r Limit	113	113	113		113	113	113
	Compre	10.0	KL/r	19.4	19.2	14.7	500	13.7	14.1	10.7
				SN	NS	NS	SLENDER	NS	NS	NS
		Web Slenderness	λr	35.88 NS	35.88 NS	35.88 NS	35.88 S	35.88 NS	35.88 NS	35.88 NS
		Web S	h/tw	12.25	12.50	18.00	42.00	15.00	19.00	33.00
		ness	λr	13.44 NS	13.44 NS	13.44 NS	12.11 NS	13.44 NS	13.44 NS	12.86 NS
		Flange Slenderness	Kc	92'0	92.0	0.76	0.62	92.0	92.0	0.70
		Flange	Kc'	1.14	1.13	0.94	0.62	1.03	0.92	0.70
			b/2tf	2.25	2.25	2.25	4.50	1.25	2.25	2.67
			(6)	32.5	33-1	1-44	1-95	46-2	46-3	72-1
37			- 50.	165						

HSS ROUND SECTIONS

22R

			Sect	tion Prop	erties					AISC Cha	apter I
(Concrete	Rei	nforcem	ent		F	Pipe		Co	mpressi	on Capacity
wt	145 pcf	fy	150	ksi	fy	46 ks	i Do	22 in	NO SLENDE	RELEMEN	NTS
f'c	8000 psi	Esr	29000	ksi	Es	29000 ks	i t	1.25 in	FyAs	3471	
					wt	490 pc	:f		C2	0.95	
		Callout	11				td	1.16	AsrEs/Ec	17.75	
Ag	299 in2	Ai	1.56	in2	Isx	64638.9 in	4 Di	19.5 in			
Ec	5098 ksi	n	2		Zx	496.8 in:	3 Dd	21.8	Pno	5852	
Ig	7098 in4	Asr	3.12	in2	ρ	0.202 OF	< Ast	75.5 in2	φPn	4389	k
		Slend	lerness C	hecks						Tensile	Capacity
		AISC	XIV Cha	pter I			- 00		Asfy	3471	k
Pipe	Slenderness	λр	λr	Max					AsrFysr	468	k
D/t	18.92 <	95	120	195	С	Compressio	n				
		57	195	195	С	Flexure			Tno	3939	
		ACI 31	8-11 Cha	pter 10					φTn	3545	k
Co	mposite Shape	Slender	ness							Flexural	Capacity
Eclg	36184968	rcomp	27.5						SLENDERNE	SS NOT C	ONSIDERED
EcAg	1522576	К	1.0								
EsIsx	1874528059	L	45						Yielding		
EsAsx	2188385	KL/r	19.7	22	ŀ	(L/r Limit			Mp=FyZ	22855	in-k
			SLEN	DERNESS	NOT	ONSIDERED			φMn	1714	ft-k

15A

			Sec	ction Pr	оре	erties			P	AISC Chapter I
Co	oncrete	Reir	nforceme	ent			Pipe		Co	mpression Capacity
wt	145 pcf	fy	150	ksi	fy	46 ksi	Do	15 in	NO SLENDE	R ELEMENTS
f'c	8000 psi	Esr	29000	ksi	Es	29000 ksi	t	1.25 in	FyAs	2295
					wt	490 pcf			C2	0.95
Ag	123 in2	Callout	11				td	1.16	AsrEs/Ec	0.00
Ac	123 in2	Ai	1.56	in2	Isx	18763 in4	Di	12.5 in		
Ec	5098 ksi	n	0		Zx	217.5 in3	Dd	14.83	Pno	3228
Ig	1198 in4	Asr	0	in2	ρ	0.289 OK	Ast	49.9 in2	φPn	2421 k
		Slende	erness Ch	ecks						Tensile Capacity
		AISC 2	XIV Chap	ter I					Asfy	2295 k
Pipe S	lenderness	λρ	λr	Max					AsrFysr	0 k
D/t	12.90 <	95	120	195	С	Compression				
		57	195	195	С	Flexure			Tno	2295
		ACI 318	-11 Chap	ter 10					φTn	2066 k
Com	posite Shap	e Slende	rness				1			Flexural Capacity
Eclg	6109839	rcomp	18.6						SLENDERNE	SS NOT CONSIDERED
EcAg	625648	К	1.0							
EsIsx	544119416	L	25						Yielding	
EsAsx	1447008	KL/r	16.1	22		KL/r Limit			Mp=FyZ	10006 in-k
	X.	-00	SLENDE	RNESS	NO	T CONSIDERED			φMn	750 ft-k

			S	ection Pr	opertie	25			AISC	C Chapte	rl
C	Concrete	Rei	nforcem	ent		Pi	pe		Comp	oression	Capacity
wt	145 pcf	fy	150	ksi	fy	46 ksi	Do	15 in	NO SLENDER	RELEMEN	ITS
f'c	8000 psi	Esr	29000	ksi	Es	29000 ksi	t	1 in	FyAs	1872	
					wt	490 pcf			C2	0.95	
Ag	133 in2	Callout	11				td	0.93	AsrEs/Ec	0.00	
Ac	133 in2	Ai	1.56	in2	Isx	15865 in4	Di	13 in			
Ec	5098 ksi	n	0		Zx	180.7 in3	Dd	14.86	Pno	2881	
Ig	1402 in4	Asr	0	in2	ρ	0.235 OK	Ast	40.7 in2	φPn	2161 k	(
		Slen	derness	Checks					Te	nsile Cap	pacity
		AIS	C XIV Cha	apter I					Asfy	1872 k	c c
Pipe	Slenderness	λр	λr	Max	224				AsrFysr	0 k	
D/t	16.13 <	95	120	195	С	Compression					
		57	195	195	С	Flexure			Tno	1872	
		ACI 3	18-11 Ch	apter 10			1		φTn	1685 k	(
Co	mposite Shap	e Slender	ness				1		Fle	exural Ca	pacity
Eclg	7147648	rcomp	18.7						SLENDERNES	SS NOT C	ONSIDERED
EcAg	676700	K	1.0								
Eslsx	460092980	L	25	S					Yielding		
EsAsx	1180272	KL/r	16.0	22		KL/r Limit	1		Mp=FyZ	8314 i	n-k
			SLEN	NDERNES	SNOT	CONSIDERED			φMn	624 f	t-k

PROPOSED ADDITIONAL MEMBERS PG56-1

Ma	terial Prope	rties			Sect	ion Prop	erties				Flexure Des	ign	
fy	50	ksi	Single	Flange	٧	Veb		Total		Flange Co	mpactness		
E	29000	ksi	А	240	Α	81	Α	561	in2	B/2tf	λρ	λr	
G	11200	ksi	х	23	х	0.563	lxx	128960	in4	1.20	9.2	24.1	С
dens	490	PCF	Ax	5520	Ax	45.56	rx	15.2	in	Web Com	pactness		
	0.284	pci	Ax2	126960			Zx	11085.6	in3	hw/tw	λρ	λr	
Pl	late Propert	ies	Ixo	2000			,,,,,,			10	90.6	137.3	С
Lb	20	ft	у	6			lyy	20160	in4	SLENDER	NESS NOT C	ONSIDERE	0
D	231000	in	Ay	1440			ry	6.0	in				
В	7,0000	in	Ay2	8640					202	Limit Stat	e 1	Yielding	
tf		in	lyo				J	16136.7			Мр	554278.1	in-
tw	9000958			- 15	1		100	201000000000000000000000000000000000000	-	Limit Stat	Branch St.	LTB	10000
							Weight	1909	PLF		Lb	240	in
hw	36	in									Lp	254.1	in
ho										LTB DOES	NOT APPLY		
3000			Í										
		Compr	ession De	esign						φMn	41570.86	ft-k	
	Flange Sler									1000			
	B/2tf	K'c	Kc	λr						-	Tension Des	ign	
	100000000000000000000000000000000000000	1.000	0.76	13.4	NS					Yielding			
	Web comp									φTn	25245	k	i
	hw/tw	λr								-5//			
		35.9	NS							RU	PTURE MUS	T BE	
	SLENDERN			DERED	ii.						CONSIDERE		
	Limit State	1	Flexural	Buckling			<u> </u>				Shear Desi	gn	ur.
		K	1.0							200000000000000000000000000000000000000	Stiffeners		
		L	240							hw/tw	hw/tw Lim	iit	
		KL/r	15.8	113.4	KL/r L	imit				16		1	
										NO STIFF	ENERS REQU	JIRED	
		Fe	1142		E3-2					kv	5		
		Fcr	49.1	ksi									
	Limit State	2	Torsiona	l Buckling	g					Cv = 1.0			
		Cw	1E+07							T I	w/tw Limit	59.2	
		Fe	1567	ksi							Cv	1.0	
		Fcr	49.3	ksi									
										Vn	3780		
	φPn	27541	k							φVn	3402	k	

PG56-1 EQUIVALENT CROSS SECTION

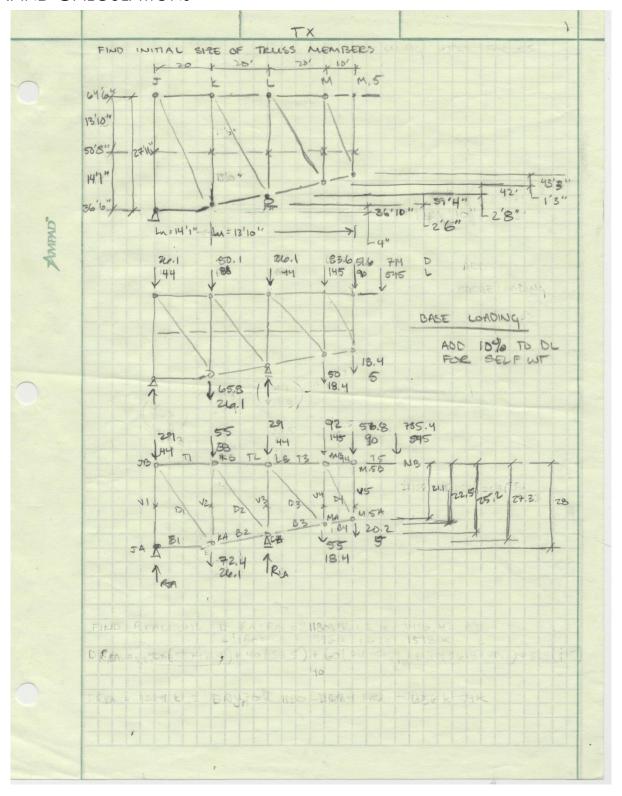
Ma	terial Prope	rties	Section Properties							Flexure Design			
fy	50 ksi		Single Flange		Web		Total			Flange Compactness			
E	29000	ksi	А	160	Α	120	А	440	in2	B/2tf	λр	λr	
G	11200	ksi	x	34	х	0.5	lxx	185813	in4	1.25	9.2	24.1	С
dens	490 PCF		Ax	5440	Ax	60	rx	20.6	in	Web Compactness			
	0.284	pci	Ax2	184960	e.		Zx	10940	in3	hw/tw	λρ	λr	
Plate Properties			Ixo	853.33						30	90.6	137.3	С
Lb	b 15 ft		у	5			lyy	9333.33	in4	SLENDERNESS NOT CONSIDERED)
D	76	in	Ау	800			ry	4.6	in				
В	20	in	Ay2	4000						Limit Stat	e 1	Yielding	
tf	8	in	Iyo	5333.3			J	6986.67			Мр	547000	in-
tw	2	in								Limit Stat	e 2	LTB	
							Weight	1497	PLF		Lb	180	in
hw	60	in				0					Lp	195.2	in
ho	68 in									LTB DOES	NOT APPLY		
Compression Design								φMn	41025	ft-k			
	Flange Slenderness			77.									
	B/2tf K'c k		Kc	λr						Tension Design			
	1.25	0.730	0.7303	13.2	NS					Yielding		7	
	Web compactness								φTn	19800	k		
	hw/tw	λr											
	30	35.9	NS							RUI	TURE MUS	TBE	
	SLENDERN	T CONSI	DERED							CONSIDERED			
	Limit State	1	Flexural	Buckling							Shear Desi	gn	
	K		1.0							Traverse S	Stiffeners		
		L	180							hw/tw	hw/tw Lim	it	
		KL/r	8.8	113.4	KL/r Li	mit				30	59.2		
										NO STIFFE	NERS REQU	IRED	
		Fe	3731	ksi	E3-2					kv	5		
		Fcr	49.7	ksi									
	Limit State 2 Torsional Buc			al Buckling	ng					Cv = 1.0			
		Cw	1E+07	in6						h	w/tw Limit	59.2	
		Fe	889	ksi							Cv	1.0	
		For	48.8	ksi									
										Vn	4560		
	φPn	k							φVn	4104	k		

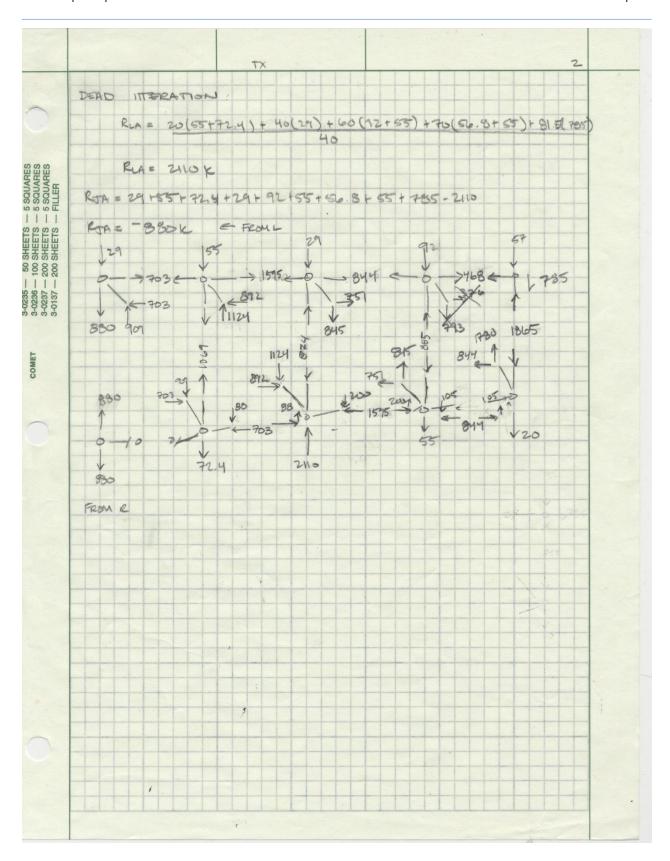
24R-1

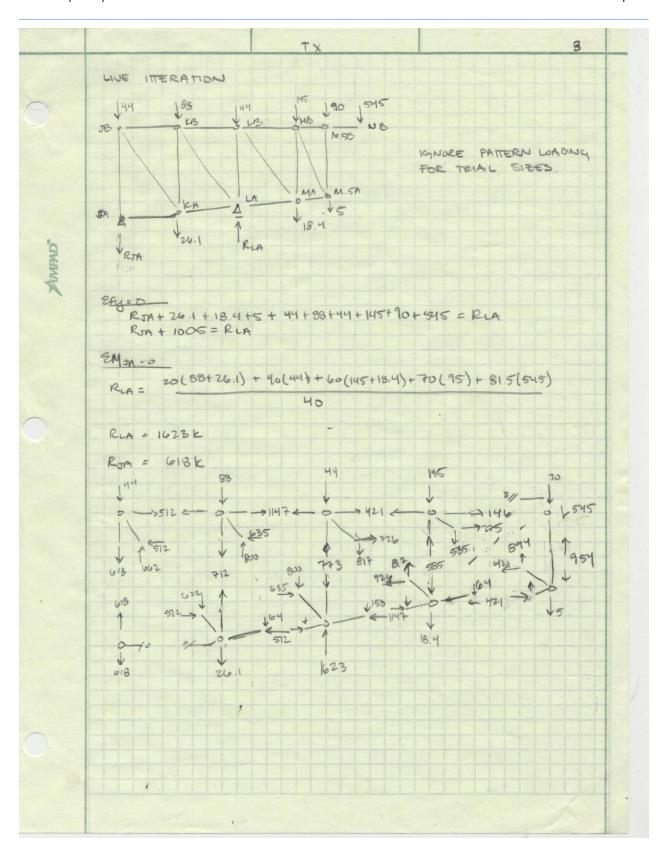
				Section F	ropert	ies			A	ISC Chapte	rl
(Concrete	Rei	nforceme	nt			Pipe		Com	pression (Capacity
wt	145 pcf	fy	150	ksi	fy	46 ks	i Do	24 in	NO SLENDE	R ELEMENT	rs
f'c	15000 psi	Esr	29000	ksi	Es	29000 ks	t	1.75 in	FyAs	5204	
					wt	490 pc	f		C2	0.95	
		Callout	11				td	1.63	AsrEs/Ec	103.69	
Ag	330 in2	Ai	1.56 i	n2	Isx	111388.7 in	4 Di	20.5 in			
Ec	6981 ksi	n	16		Zx	798.3 in:	3 Dd	23.755	Pno	11030	
Ig	8669 in4	Asr	24.96 i	n2	ρ	0.255 Ok	Ast	113.1 in2	φPn	8272 k	¢
		Sle	nderness	Checks					T	ensile Cap	acity
		Al	SC XIV Ch	apter I					Asfy	5204 k	C
Pipe	Slenderness	λp	ir 1	Max					AsrFysr	3744 k	C
)/t	13.71 <	95	120	195	С	Compressio	n				
		57	195	195	С	Flexure			Tno	8948	
		ACI	318-11 Ch	apter 10					φTn	8053 k	c
Co	omposite Sha	pe Slender	ness						FI	exural Cap	oacity
clg	60520963	rcomp	29.4						SLENDERNE	SS NOT CO	NSIDERED
cAg	2304189	K	1.0								
slsx	3230271344	L	25						Yielding		
sAsx	3280962	KL/r	10.2	22		KL/r Limit			Mp=FyZ	36722 i	n-k
			SLEN	DERNESS	NOTC	ONSIDERED			φMn	2754 f	t-k

APPENDIX D: ETABS VERIFICATION

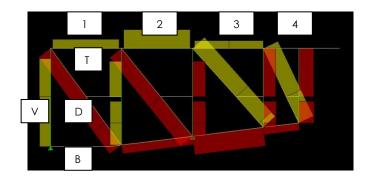
HAND CALCULATIONS







MEMBER SELECTION TABLE



						Truss	Loads							Tri	al Men	ber Se	lection	n
		L	Pdx	Pdy	Plx	Ply	Pd	PL	1.2Pd	1.6PI	Pu	T/C	p max	L	р	Trial S	ize	φPn
D	1	33.8	703	909	512	662	1149	837	1379	1339	2718	С	0.368	18	0.325	W14x	283	3077
	2	32.2	892	1124	635	800	1435	1021	1722	1634	3356	С	0.298	18	0.266	W14x	342	3759
	3	29.7	751	845	726	817	1130	1093	1357	1749	3105	Т	0.322	0	0.271	W14x	342	3690
	4	23.3	844	1780	421	894	1970	988	2364	1581	3945	Т	0.253	0	0.204	W14x	455	4902
Т	1	20	703		512		703	512	843.6	819.2	1663	T	0.601	0	0.521	W27x	178	1919
	2	20	1595		1147		1595	1147	1914	1835	3749	Т	0.267	0	0.172	W27x	539	5814
	3	20	844		421		844	421	1013	673.6	1686	Т	0.593	0	0.575	W27x	161	1739
	4	10	468		146		468	146	561.6	233.6	795.2	Т	1.258	0	0.575	W27x	161	1739
В	1	20.05	0		0		0	0	0	0	0	С	0.597	22	0.552	W27x	217	1812
	2	20.16	703	88	512	64	708	516	850.2	825.6	1676	С	0.597	22	0.552	W27x	217	1812
	3	20.16	1595	200	1147	150	1607	1157	1929	1851	3780	С	0.265	22	0.205	W27x	539	4878
	4	10.08	844	105	421	64	851	426	1021	681.3	1702	С	0.588	12	0.54	W27x	161	1852
٧	1	28	880		618		880	618	1056	988.8	2045	Т	0.489	0	0.482	W14x	193	2075
2417	2	27.3	1069		712		1069	712	1283	1139	2422	Т	0.413	0	0.399	W14x	233	2506
	3	25.2	874		773		874	773	1049	1237	2286	С	0.438	14	0.406	W14x	211	2463
	4	22	885		585		885	585	1062	936	1998	С	0.501	14	0.487	W14x	176	2053
	5	21.1	1865		954		1865	954	2238	1526	3764	С	0.266	14	0.247	W14x	342	4049

APPENDIX E: TRUSS DESIGN CALCULATIONS

EXTERIOR WALL ALLOCATION

Truss 0.9

								Truss 0	.9								
Wt	15	PSF	73	3		Н		ı		J		K		L		L.5	
Ĺ	Elev.	Ht	W	Wt	W	Wt	W	Wt	W	Wt	W	Wt	W	Wt	W	Wt	
9	151.67	9.1	20	2.73	20	2.73	11.5	1.57							av K		
8	133.5	19.9	20	5.98	20	5.98	20	5.98	17.1	5.11	8.67	2.59					
7	111.83	22.7	20	6.80	20	6.80	20	6.80	20	6.80	20	6.80	14.33	4.87		5 1.	70
6	88.17	23.7	20	7.10	20	7.10	20	7.10	20	7.10	20	7.10	14.33	5.09		5 1.	78
	L	oads A	pplied	22.6	k	22.6	k	21.4	k	19.0	k	16.5	k	10.0	k	3	.5
L	Elev.	Ht.	KLF														
5	64.5	11.8	0.18														

Truss X

				1	Truss X				
	A	REA	Н	Į Į	J	K	L	M	M.5
L		Elev.	20	40	60	80	100	120	130
	5	64.5	138.3	138.3	138.3	138.3	138.3	172.9	34.6
	4	50.67	283.4	283.4	283.4	283.4	283.4	283.4	283.4
	3	36.5	245.0	245.0	219.6	165.2	110.8	56.4	29.2
	2	24.25	83.4	29.0					
		Bottom	20.08	22.8	25.52	28.24	30.96	33.68	35.04
	L	DAD	Wt =	15	PSF				
L		Elev.	Н	Į.	J	K	L	M	M.5
1	5	24.25	2.1	2.1	2.1	2.1	2.1	2.6	0.5
	4	50.67	4.3	4.3	4.3	4.3	4.3	4.3	4.3
	3	36.5	3.7	3.7	3.3	2.5	1.7	0.8	0.4
	2	24.25	1.3	0.4					

TRUSS 0.9

Level 6 Loads

					Z*-G		De servicione								J	.5			To a	
evel		Area	DL	DW	DC	LL	SL	D	L	S	Level	Area	DL	DW	DC	LL	SL	D	L	S
	9	520	84	2.73	2	50	22	48.4	26.0	11.4	6	390	168	0	0	150	C	65.5	58.5	0
	8	590	121	5.98	2	150	0	79.4	88.5	0.0										
	7	660	121	6.80	2	100	0	88.7	66.0	0.0					W	*-K				
	6	390	148	7.10	2	130	4.4	66.8	50.7	1.7	Level	Area	DL	DW	DC	LL	SL	D	L	S
	-110					T	otal =	283	231	13	8	245	187	2.59	2	100	C	50.4	24.5	0
											7	520	146	6.80	2	100	C	84.8	52.0	0
					G.5						6	390	170	7.10	2	150	C	75.2	58.5	0
Level		Area	DL	DW	DC	LL	SL	D	L	S							Total =	210	135	32
	6	390	152.1	0	0	130	4.4	59.3	50.7	1.7										Ĩ
															K	.5	2 1			-
					W*-H	į.					Level	Area	DL	DW	DC	LL	SL	D	L	S
.evel		Area	DL	DW	DC	LL	SL	D	L	S	6	390	173	0	0	150	C	67.4	58.5	0
	9	470	84	2.73	2	50	22	44.2	23.5	10.3										î
	8	530	121	5.98	2			72.1	0.0	0.0					W	*-L		8 1		
	7	580	121	6.80	2	100		79.0	58.0	0.0	Level	Area	DL	DW	DC	LL	SL	D	L	S
	6	390	156	7.10	2	130	4.4	70.0	50.7	1.7	7	370	183	4.87	2	100	C	74.6	37.0	0
	- 111					T	otal =	265	132	12	6	364	176	5.09	2	150	C	71.2	54.6	0
											0	0 0					Total =	146	92	3
					H.5															
Level		Area	DL	DW	DC	LL	SL	D	L	S					W*	-L.5				
	6	390	157.8	0	0	140	2.2	61.5	54.6	0.9	Level	Area	DL	DW	DC	LL	SL	D	L	S
											7	150	183	1.70	2	100	C	31.1	15.0	0
					W*-I						6	372	183	1.78	2	150	C	71.9	55.9	0
Level		Area	DL	DW	DC	LL	SL	D	L	S							Total =	103	71	de —
	9	280	84	1.57	2	50	22	27.1	14.0	6.2										Ĩ
	8	500	154	5.98	2	125	0	85.0	62.5	0.0					W.	*-M	2			80
	7	550	121	6.80	2	100	0	75.3	55.0	0.0	Level	Area	DL	DW	DC	LL	SL	D	L	S
	6	390	159	7.10	2	150	0	71.3	58.5	0.0	6	411	203	0.00	2	150	C	85.5	61.7	0
	- 110					Т	otal =	259	190	6										Î
	_]]														N	1.5	2 1			
					1.5						Level	Area	DL	DW	DC	LL	SL	D	L	S
Level		Area	DL	DW	DC	LL	SL	D	L	S	6	390	203	33.00	2	150	C	114.2	58.5	0
	6	390	162.8	0	0	150	0	63.5	58.5	0.0										
					W*-J															
Level		Area	DL	DW	DC	LL	SL	D	L	S										
	8	365	187	5.11	2	100	0	75.4	36.5	0.0										
	7	530	121	6.80	2	100	0	72.9	53.0	0.0										
	6	390	166	7.10	2	150	0	73.9	58.5	0.0										
						Т	otal =	222	148	0										

Level 5 Loads

Load In	format	ion
Wall Load	0.18	KLF
Dead Load	109	PSF
Live Load	200	PSF
Grid Angle	9	deg
	0.157	rad
Trib Angle	0.079	rad

			X Trib			ΥT	rib		Lo	ad Con	npon	ent	s	Loa	ds
Grid	Х	Start	End	Dx	Base	Start	End	Dy	At	AD	PW	PC	AL	PD	PL
F	0	0	5	5	0.31	0.31	0.70	0.39	2.5	0.3	3.6	2	0.5	5.8	0.5
F.5	10	5	15	10	1.09	0.70	1.49	0.79	10.9	1.2			2.2	1.2	2.2
G	20	15	25	10	1.88	1.49	2.27	0.79	18.8	2.0	3.6	2	3.8	7.6	3.8
G.5	30	25	35	10	2.67	2.27	3.06	0.79	26.7	2.9			5.3	2.9	5.3
H	40	35	45	10	3.45	3.06	3.85	0.79	34.5	3.8	3.6	2	6.9	9.3	6.9
H.5	50	45	55	10	4.24	3.85	4.63	0.79	42.4	4.6			8.5	4.6	8.5
I.	60	55	65	10	5.03	4.63	5.42	0.79	50.3	5.5	3.6	2	10.1	11.0	10.1
1.5	70	65	75	10	5.81	5.42	6.21	0.79	58.1	6.3			11.6	6.3	11.6
J	80	75	85	10	6.60	6.21	7.00	0.79	66.0	7.2	3.6	2	13.2	12.7	13.2
J.5	90	85	95	10	7.39	7.00	7.78	0.79	73.9	8.1			14.8	8.1	14.8
K	100	95	105	10	8.18	7.78	8.57	0.79	81.8	8.9	3.6	2	16.4	14.5	16.4
K.5	110	105	115	10	8.96	8.57	9.36	0.79	89.6	9.8			17.9	9.8	17.9
L	120	115	125	10	9.75	9.36	10.14	0.79	97.5	10.6	3.6	2	19.5	16.2	19.5
L.5	130	125	135	10	10.54	10.14	10.93	0.79	105.4	11.5			21.1	11.5	21.1
M	140	135	145	10	11.32	10.93	11.72	0.79	113.2	12.3	3.8	2	22.6	18.2	22.6
M.5	150	145	155.75	10.8	12.11	11.72	12.56	0.85	130.5	14.2			26.1	14.2	26.1
N.2	162	155.8	161.5	5.75	13.02	12.56	13.02	0.45	73.5	8.0	2.0	2	14.7	12.1	14.7

Reactions

	Reactions (k)
Level 6	Н	N.2
D	0	340
L	0	244
S	0	0.5
Level 5		
D	1112	404
L	747	295
S	18	0.5

TRUSS N.2

	Load Pro	perties	
	Level 6	Level 5	
Trib W	6.5	6.5	ft
DL	203	109	PSF
LL	150	200	PSF
1	Distribute	ed Loads	
wD	1.3	0.7	klf
wL	1.0	1.3	klf
	Point I	oads	
Pd	340	404	k
Pl	244	295	k
	R	eactions	
	8	N.2	6
	D	1340	-350
	L	945	-252
	S	2	-1
	7.7	73	0

Truss X Load Summary

Le	evel	J	J.5	K	K.5	L	L.5	M	M.5	N.2
5	D	2.1	24.0	26.1	24.0	2.1	32.7	50.9	51.6	1340.0
0.00	L		44.0	44.0	44.0		60.0	85.0	90.0	945.0
	S									2.0
4	D	17.5		31.7		22.6		45.9	16.9	
	L	6.8		14.0		10.3		18.4	5.0	
3	D	19.2		34.1		9.1		3.6	1.5	
	L	6.8		12.1		9.0		0.0	0.0	

Load	Reactio	on (k)
	J	L
D	-1072	3392
L	-787	2419
S	-2	4
Pu	-2547	7943

AREA TAKEOFF

Area DL DW DC LL D Level Area DL DW DC LL D Level Area DL DW DC LL D C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C C <th< th=""><th>H-X</th><th>#X </th><th>×-</th><th></th><th>3 40</th><th></th><th></th><th></th><th></th><th>-</th><th></th><th>X-J.5</th><th>5.</th><th></th><th></th><th></th><th></th><th></th><th></th><th></th><th> <u>~</u> </th><th></th><th></th><th></th></th<>	H-X	# X	×-		3 40					-		X-J.5	5.								<u>~</u>			
120 98 0 0 240 440 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540 540	DL DW DC LL D	DW DC LL D	DC [[D	11 0	0		_		Level	Ā		DW	DC	1	٥	7	Le	Ā	۵	0	20			٦
120 98 0 0 50 11.8 6.0 4 250 98 4.3 2 50 34.4 1 1 1 1 1 1 1 1 1	170 20 1.3 0.5 0 5.2 0.0	1.3 0.5 0 5.2	0.5 0 5.2	0 5.2	5.2	_	0.0			× 1.20		0	0	200	24.0	44.0		-			2349	7550	27	85.0
125 118 0 0 5 14.8 6.3 4(3) 350 15 0.0 0 0 0 5.3												0	0	20	11.8	0.9		******			0.00			12.5
X-K Area DL LL DL LL CL	X-H.5	X-H.5	X-H.5	9				6 1				0	0	20	14.8	6.3	4	ecau:						0.0
Area DL DW DC LL D Level Area DL DN DC LL D Level Area DL DN DC LL D LL	Area DL DW DC LL D L	DW DC LL	DC II	п		D L												853.15					VOECE SITURE	0.0
Area DL LL D LL Curl Level Area DL DN DC LL D 130 98 4.3 2 50 19.0 6.5 5 450 109 0.5 2 2 0 5.16 135 115 2.5 2 100 13.5 4 100 98 4.3 2 50 51.6 135 115 2.5 100 13.5 44 100 98 4.3 2 50 16.9 4rea DL N 2 44 100 98 4.3 2 50 16.9 220 109 D L 44 100 98 4.3 0 0 1.3 220 109 D L D L B L B L B L B 1.3 4120 98 0 0 150 <td>220 109 0 0 200 24.0 44.0</td> <td>0 0 200 24.0</td> <td>0 200 24.0</td> <td>200 24.0</td> <td>24.0</td> <td></td> <td>44.0</td> <td>6-6</td> <td></td> <td>8</td> <td></td> <td>×</td> <td>×</td> <td>8</td> <td>20</td> <td></td> <td></td> <td></td> <td></td> <td></td> <td></td> <td></td> <td></td> <td></td>	220 109 0 0 200 24.0 44.0	0 0 200 24.0	0 200 24.0	200 24.0	24.0		44.0	6-6		8		×	×	8	20									
220 109 2.1 0 200 26.1 44.0 Level Area DL DW DC LL D 130 98 4.3 2 50 13.5 450 109 0.5 2 20 51.6 135 115 2.5 2 100 13.5 44 100 98 4.3 2 50 51.6 Area DL N 2 10 20.0 13.5 44.0 15 0.4 0.5 0 0 1.3 Area DL DW DC LL D L 44.0 1.5 0.4 0.5 0 0 1.5 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	170 98 0 0 50 16.7 8.5 L	0 0 50 16.7 8.5	0 50 16.7 8.5	50 16.7 8.5	16.7 8.5	8.5			eve	2/6		DW	DC	П	D	1				^	(-M.5			
130 98 4.3 2 50 13.5 6.5 450 100 98 4.3 2 200 51.6 135 115 2.5 2 100 13.5 4 100 98 4.3 2 50 51.6 Area DL N L D L N 4(3) 85 15 0 0 0 1.3 Area DL DN DC LL D L 0 0 0 1.3 115 98 0 0 20 24.0 44.0 1.5 1.2 1.2 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 1.5 <	200 118 0 0 50 23.6 10.0	0 0 50 23.6	0 50 23.6	50 23.6	23.6		10.0	6 9	100			2.1	0	200	26.1	44.0	Lev			DW		=	O	-
135 115 2.5 2 100 13.5 4 100 98 4.3 2 50 16.9 X-X-X-S Area DL DW DC LL D L A4.0 15 0.4 0.5 0.1 1.3 220 109 0 0 20 24.0 44.0 3 40 1.5 0 0 1.5 115 98 0 0 50 13.4 44.0 3 4 1.5 0 0 1.5 112 98 0 0 50 13.4 44.0 3 4 1.5 1.5 1.2 1.2 1.5 1.2 1.2 1.5 1.2 1.2 1.2 1.2 1.2 1.2 1.2 1.2 1.2 1.2 1.2 1.2 1.2 1.2 1.2 1.2 1.2 1.2 1.2 1.2 1.2 1.2 1.2 1	80 20 0.0 0.5 0 2.1 0.0	0.0 0.5 0 2.1	0.5 0 2.1	0 2.1	2.1		0.0					4.3	2	50	19.0	6.5		0,000				-151.01	225	90.0
Afrea DL DW DC LL D L Afrea DL L Afrea Afrea DL LL DL LL D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L D L </td <td></td> <td></td> <td></td> <td></td> <td></td> <td></td> <td></td> <td>w 19</td> <td>200</td> <td>200</td> <td>25 26</td> <td>2.5</td> <td>2</td> <td>100</td> <td>20.0</td> <td>13.5</td> <td></td> <td></td> <td></td> <td></td> <td></td> <td></td> <td></td> <td>5.0</td>								w 19	200	200	25 26	2.5	2	100	20.0	13.5								5.0
Area DL DN DC LL D L A.0 T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T T	X-I	Y-1	X-1														4	(3)			-		Parallel I	0.0
Area DL DW DC LL D 220 109 0 200 24.0 115 98 0 50 13.7 85 112 0 150 9.5 120 112 0 0 150 9.5 Area DL DW DC LL D 300 109 0 200 32.7 235 98 0 0 50 23.0 95 15 0 0 0 1.4	Area DL DW DC LL D L	DW DC II	DC II	П	- 0	1 Q	1	S. 33				X-X	(.5	2 3				က					2000	0.0
220 109 0 0 200 24.0 115 98 0 0 50 13.7 85 112 0 0 150 9.5 120 112 0 0 150 9.5 Area DL DW DC LL D 300 109 0 0 200 32.7 235 98 0 0 50 23.0 95 15 0 0 1.4	220 109 2.1 0 200 26.1 44.0 Level	2.1 0 200 26.1 44.0	0 200 26.1 44.0	200 26.1 44.0	26.1 44.0	44.0		Leve		26.54		DW	DC	п	D	Г								
115 98 0 0 50 13.7	160 98 4.3 2 50 21.9 8.0	4.3 2 50 21.9 8.0	2 50 21.9 8.0	50 21.9 8.0	21.9 8.0	8.0					S 97	0	0	200	24.0	44.0								
85 112 0 0 150 9.5 120 112 0 0 150 13.4 X-L.5 Area DL DW DC LL D 300 109 0 200 32.7 235 98 0 50 23.0 95 15 0 0 1.4	215 118 3.7 2 50 33.3 10.8	3.7 2 50 33.3	2 50 33.3	50 33.3	33.3		10.8		- 7		.5-24	0	0	20	13.7	8.9								
120 112 0 0 150 13.4 X-L.S Area DL DW DC LL D 300 109 0 200 32.7 235 98 0 0 50 23.0 95 15 0 0 1.4	160 20 0 0 0 3.2 0.0	0 0 0 3.2 0.0	0 0 3.2 0.0	0 3.2 0.0	3.2 0.0	0.0		4	(3	Washington, and		0	0	150	9.5	12.8								
Area DL DW DC LL D 300 109 0 200 32.7 235 98 0 0 50 23.0 95 15 0 0 1.4	66 20 0.4 0.5 0 2.3 0.0	0.4 0.5 0 2.3	0.5 0 2.3	0 2.3	2.3		0.0	(a.)	1000			0	0	150	13.4	18.0								
Area DL DW DC LL D 300 109 0 200 32.7 235 98 0 0 50 23.0 95 15 0 0 1.4																								
Area DL DW DC LL D 300 109 0 200 32.7 235 98 0 0 50 23.0 95 15 0 0 1.4	X-1.5	X-1.5	X-1.5	2			8 S	16 - 3			30	1-X	5	3	95									
300 109 0 0 200 32.7 235 98 0 0 50 23.0 95 15 0 0 1.4	Area DL DW DC LL D L	DW DC IL D L	DC (I, D L	ת ס ת	1 Q		T T	Le	ve	2.5		DW	DC	п	D	Г								
235 98 0 0 50 23.0 95 15 0 0 0 1.4	220 109 0 0 200 24.0 44.0	0 0 200 24.0	0 200 24.0	200 24.0	24.0		44.0		715	-		0	0	200	32.7	0.09								
95 15 0 0 0 1.4	150 98 0 0 50 14.7 7.5	0 0 50 14.7	0 50 14.7	50 14.7	14.7		7.5			1.00		0	0	20	23.0	11.8								
	145 118 0 0 50 17.1 7.3	0 0 50 17.1	0 50 17.1	50 17.1	17.1		7.3	0-12			300	0	0	0	1.4	0.0								
	50 20 0 0.5 0 1.5 0.0	0 0.5 0 1.5	0.5 0 1.5	0 1.5	1.5		0.0																	

Truss J

			Area Tak	eoff			
	Dead	Loads	40		4		0.9
	w	D	wD (klf)	Α	PD	PC	PD
5	10	109	1.1	445	48.5	0.0	1072
4	10	98	1.0	445	43.6	2.0	
3	10	118	1.2	445	52.5	2.0	
	Live	Loads	22		4		0.9
	w	D	wD (klf)	Α	PL	PC	PL
5	10	200	2.0	445	89.0		787
4	10	50	0.5	445	22.3		
3	10	50	0.5	445	22.3		
		Level 5			Rea	actions	(k)
Ì	W	0.9	X	4		3	4
D	1.1	1072	-18.7	48.5	D	-973	429
L	2.0	787	-6.8	89.0	L	-22	179
S			-4		S	-6	2
		Level 4				250	
D	1.0		-361.7	45.6			
L	0.5		283.6	22.3			
		Level 3					
D	1.2		-419.7	54.5			
L	0.5		-333.9	22.3			

Truss H

		Level 5					
	0.9	X	4	W	- 8	Reacti	ons (k)
D	1112	-19	35	2.2	Load	3	4
L	747	-7	64	4	D	2479	-906
S	18				L	1661	-572
		Level 4	70		s	34	-16
	0.9	Х	4	w	Pu	5646	-2008.8
D		780	31	2.0	Colum	24R-1	15A
L		568	19	1			
		Level 3					
	0.9	Х	4	w			
D		-640	38	2.36			
L		-464	32	1			

APPENDIX F: FOUNDATION IMPACT

Ī		8 3		8 3		10				0 0		8 3				10		8 0		
		Point Loads	50 k	50 k	0 K	Design Loads	-293 k	-202 k	-1 k			Point Loads	104 k	105 k	0 k	Design Loads	3516 k	2524 k	4 k	
		Po	Pd1	PI1	Ps1	Des	Pd2	PI2	Ps2			Po	Pd1	Pl1	Ps1	Des	Pd2	PI2	Ps2	
		6	250 sft	198 PSF	200 PSF	r	0 sft	126 PSF	100 PSF			i.	526 sft	198 PSF	200 PSF	r	0 sft	126 PSF	100 PSF Ps2	
	6-N.2	Exterior	25	19	20	Interior		12	10		3-F	Exterior	52	19	20	Interior		12	10	
			Area	DI	П		Area	7	П				Area	DI	П		Area	DI	П	
		1	350 k	.252 k	-1 k	8 K			-675 k				2 k	2419 k	4 k	20 k			8260 k	
		Loads In	-35	-25	0000	200			-67			Loads In	3392	241	20083	2				
		_	Pd	PI	Ps	Self			= nd				Pd	Ы	Ps	Self			= nd	
		ds	×	¥	0 k	ds	¥	¥	2 k	00 - 50 82 - 1		ds	×	¥	0 k	ds	¥	¥	×	
		Point Loads	74 K	64 k	0	Design Loads	511 k	243 k	7			Point Loads	104 k	105 k	0	Design Loads	-861 k	-74 k	-6 K	
		Poi	Pd1	PI1	Ps1	Desi	Pd2	PI2	Ps2			Poi	Pd1	PI1	Ps1	Desi	Pd2	PI2	100 PSF Ps2	
			112 sft	198 PSF	200 PSF	2. 3	414 sft	126 PSF	100 PSF				526 sft	198 PSF	200 PSF	2.3	0 sft	126 PSF	PSF	
	4-)	Exterior	112	198	200	Interior	414	126	100		3-J	Exterior	526	198	200	Interior	0	126	100	
		E	Area	DL	П	4	Area	DI	11			E	Area	DL	11	-	Area	DL	П	
			k	179 k	2 k	8 k	117.		Section 15				k	¥	-6 k	8 k		-	¥	
		Loads In	429 k	179	2	8			1003 k			Loads In	-973	-179	9-	8			-1154 k	
		Z	Pd	Ы	Ps	Self			= nd			C	Pd	Ы	Ps	Self		-	= nd	-
		S	×	×	×	ds	×	×	×	(I)		S	×	×	×	ds	×	×	×	
		t Loads	70 k	58 k	0 k	gn Loads	-820 k	-514 k	-16 k			t Loads	100	100	y 0	gn Loads	2587 k	1761	34 k	
		Poir	Pd1	PI1	Ps1	Desig	Pd2	PI2	Ps2			Poin	1pd	Pl1	Ps1	Desig	Pd2	PI2	Ps2	
				198 PSF PI1		0.00		201 20		2			0.00	03 23	_	0.00		PSF		
2	4-H	Exterior	56 sft	198	200 PSF	Interior	470 sft	126 PSF	100 PSF	8	3-H	Exterior	470 sft	198 PSF	200 PSF	Interior	56 sft	126 PSF	100 PSF	
	-5.0	Ext	Area	L		Int	Area	DL		8		Ext	Area	7		Int	Area	DL		
i				k DL	k LL	k	A	0	k L			0.0		k DL	k LL	k	A	O	k LL	
		Loads In	-898 k	-572 k	-16 k	8 k			-1814 k			Loads In	2479 k	1661 k	34 k	8 k			5938 k	
		Lo	Ьd	PI	Ps	Self			= nd			Lo	Ьd	PI	Ps	Self			= nd	

APPENDIX G: DEFLECTION CALCULATIONS

	Tr	Truss In	form	nformation				-	Itemized	P	Live	Load [Live Load Deflection	tion	Tota	Total Load Deflection	Deflect	tion
		Lsin	Lsimple	SI	LC	Lcant	Lc	Def	Deflections (in)	(in)	Support	ort	/piW		Support	port	/piw	
I.D.	Condition ft	30000	in	in	ft	in	in	O	1	T	1	2	End	Max	1	2	End	Max
1	Cantilever	0	0	0	9	4	76	-0.46	-0.03	-0.49	0.00	0.00	0.03	0.03	0	0	0.49	0.49
×	Cantilever	40	9	486	42	-	505	2.12	1.50	3.62	0.03	0.00	1.50	1.53	0.49	0	3.62	4.13
N.2	Cantillever	43	11	527	26	-	313	0.58	0.41	0.99	0.00	1.53	0.41	2.85	0	4.129	0.99	7.57
I	Cantilever	0	0	0	19	9	234	1.13	0.76	1.89	0.00	0.00	0.76	0.76	0	0	1.89	1.89
6.0	Simple	121	9	1458	0	0	0	1.21	0.875	2.09	0.76	2.85	0.88	2.68		1.89 7.572 2.085	2.085	6.82
6:0:X	Simple	13	0	0 156 0 0	0	0		1.05	0.75	0 1.05 0.75 1.80	0.03	0.03 1.80	ě	1.83	0.49	1.83 0.49 6.82	7	7.31

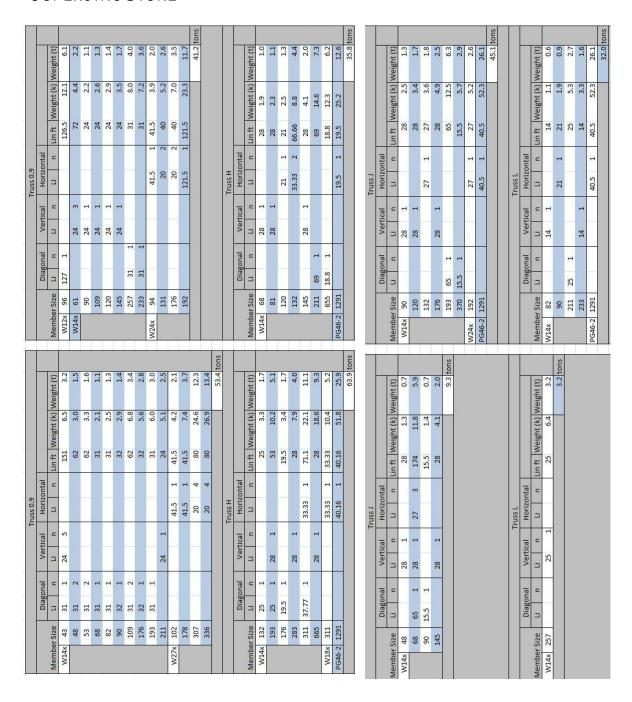
	Allo	wable	Defle	ction	Allowable Deflection at Cantilever	ilever	
	×	\	Xp	dγ	_	1/180	/180 L/120
3-r	2664	222		-234	498 -234 550.2	3.06	4.59
0.9-N.2 3162	3162	-12		0	6 0		
	A	lowab	le Def	lectio	Allowable Deflection for X:0.9	6.0:	
	×	λ	Хþ	λþ	7	1/360 1/240	L/240
3-1	2184	222	2 70	0 -234	234	0.65	0.98
f-6.0	2184	-12					

APPENDIX H: COST AND SCHEDULE INFORMATION

RS MEANS CITY DATA

10 11 10	0 9 9 9				4		
	1 1 1	7 8 75 76 82 82	80 81 77 79 102	96. 94. 95. 97. 95. 94.	97.4 96.0 99.1 91.2 94.4 95.7 94.8 95.3 91.7	91.9 90.6 89.2 92.3 89.7 90.5 91.8 90.2 89.1	91.2 87.3 85.3 87.8 89.5 89.1 85.3
119.9 122.0 110.3	106.1 103.2 92.8 103.9 119.4 103.7 97.9 99.8 94.5	54.9 64.3 57.3 54.4 63.6 65.3	58.1 59.5 54.0 56.2	94.5 91.8 89.4 95.2 89.5 89.9	108.2 94.3 95.2 102.9 81.7 91.7 93.2 92.4 95.7	75.4 75.1 76.8 75.0 75.0 75.0 9 77.5 9 77.0 9	73.0 8: 70.6 8: 74.0 8: 77.8 89 78.1 89
97.2 100.4 101.4 99.7	100.1 97.3 96.0 95.6 97.2 97.1 99.1 98.4 98.0	96.1 99.9 97.7 97.3 97.4 96.5	97.5 98.1 96.1 97.4	98.2 97.1 100.3 99.1 99.5 97.5 97.0	100,7 99,9 96,6 96,2 98,6 96,6 97,7 96,7 95,1 98,6	103.3 102.5 100.3 104.5 101.4 102.7 103.1 100.6 99.3	100.4 98.6 96.8 98.6 98.7 97.8 97.6 98.1
untain ul olis er	ets Falls	sale ille wood n	el iomb nibus	ng Green bal ille ille ille Girardeau on Bluff	s City eph the nville on City is	s	nd

SUPERSTRUCTURE



					tons																			tons				
		Lin ft Weight (k) Weight (t)	7.0	26.2	26.2 tons									Lin ft Weight (k) Weight (t)	0.1	6.0	1.3	0.3	6.0	1.2	6.0	2.1	11.9	19.3 tons				
		t (k) W	14.1	52.4										t (k) W	0.2	1.7	2.5	9.0	1.7	2.4	1.7	4.1	23.7					
		Weigh												Weigh						70.0	-							
		Lin ft	70	70										Lin ft	10.13	28	28	20.25	20.25	40.5	20.25	30.38	70.88					
2	ontal	u	1	1									ontal	u	0.5			1	1	2	1	1.5	3.5					
Truss N.2	Horizontal	ŋ	70	70								Truss X	Horizontal	ij	20.25			20.25	20.25	20.25	20.25	20.25	20.25					
	Vertical	u										<i>y</i>	Vertical	п		-	1											
	Ver	Li											Ver	'n		28	28											
	Diagonal	u											Diagonal	c														
	Dia	П											Dia	בי	2	-	0	1	4	0	4	5	2					
		Member Size	201	3 748										Member Size	15	61	90	31	84	9	84	135	335					
		Mem	W38x	PG46-3										Mem	W10x	W14x		W16x		W18x	W27x	W36x	W44x					
											ons																	ons
		Veight (t)	1.2	1.4	2.9	3.1	2.8	4.4	7.4	13.0	36.0 tons	*		Veight (t)	1.7	2.7	3.6	4.0	5.0	9.5	11.3	4.8	0.7	1.2	10.9	16.5	49.7	121.6 tons
		Weight (k) Weight (t)	2.3	2.9	5.7	6.1	5.6	8.7	14.8	25.9				Lin ft Weight (k) Weight (t)	3.4	5.4	7.3	8.0	10.0	18.5	22.7	9.7	1.3	2.5	21.8	33.0	99.5	
		Lin ft W	48	24	36	29	24	34	70	70				Lin ft W	20	33.82	34.5	34.5	32.2	24	57	21.25	10.2	17	40.5	44.18	42	
200	ontal	u							1	1			ontal	u									1	-	2	1	1	
Truss N.2	Horizontal	Ü							0/	20		Truss X	Horizontal	Li									10.2	17	20.25	44.18	42	
-	Vertical	u	2	1			1					3883	Vertical	u	2												ĺ	
	Ver	0	24	24			24						Ver	ij	52													
	Diagonal	u			1	-		1					onal	u		-	1	-	-	1	-	1						
	Diag	LÍ			36	29		34					Diagonal	:		33.82	34.5	34.5	32.2	54	57	21.25						
		er Size	48	120	159	211	233	257	211	370				ir Size	89	159	211	233	311	342	398	455	129	146	539	748	2369	
		Member Size	W14x						W18x	W24x				Member Size	W14x								W27x			PG46-3	PG56-1	

FOUNDATIONS

REINFORCEMENT

DYWIDAG Prestressing Steel Threadbar System

DYWIDAG Prestressing Steel Threadbar is a high tensile alloy steel bar which features a coarse right-hand thread over its full length. The system is proven worldwide and offers versatility in a range of applications.

Manufactured in accordance with the German Certificate of Approval (Deutsches Institut für Bautechnik), the system also offers general conformance with BS 4486: High Tensile Steel Bars for Prestressing of Concrete. During the steel making process, the threadbars are hot rolled, quenched and tempered, followed by cold working and further tempering, to achieve the necessary performance.

DYWIDAG Prestressing Steel Threadbars, 15mm - 75mmØ are suitable for all static loading applications. Additionally, for post-tensioning and dynamic applications, DYWIDAG Prestressing Steel Threadbars 26.5mm - 40mmØ, see note (c) below, offer a fatigue resistance in excess of 2 million load cycles over a tensile range of 630 - 682N/mm² as specified in the European Technical Approval No. ETA - 05/01/23 and ETAG 013. Stress relaxation when loaded to 70% fpu is less than 3.5% over a 1000 hour period in accordance with BS4486.

Key features of the system are:

- Fully threaded bar can be cut and coupled at any point.
 Coarse pitch threadform (d/2), right-hand, with two faces ensuring the thread is self cleaning, Ideal for construction site use.
 Low relaxation steel minimum relaxation during service life.
 Prestressing grade steel high strength offers weight savings and reduced working diameters.

Technical Data for Prestressing Steel Threadbar

Nominal Diameter	Steel Grade	Ultimate Strength fpu	0.1% (a) Proof Strength	70% (b) Ultimate Strength	50% Ultimate Strength	Cross Sectional Area	Diameter Over Threads	Thread Pitch	Bar Weight
mm	N/mm²	kN	kN	kN	kN	mm ²	mm	mm	kg/m
15	900/1100	195	159	136	98	177	17	10	1.44
20	900/1100	345	283	241	173	314	23	10	2.56
26.5	950/1050	579	523	405	290	551	30 .	. 13	4.48
32	950/1050	844	764	591	422	804	36	16	6.53
36	950/1050	1069	967	748	535	1018	40	18	8.27
40	950/1050	1320	1194	924	660	1257	45	20	10.21
47	950/1050	1822	1648	1275	911	1735	52	21	14.10
57	835/1035	2671	2155	1870	1335	2581	64	21	20.95
65	835/1035	3447	2771	2413	1724	3318	71	23	27.10
75	835/1035	4572	3645	3200	2286	4418	82	24	35.90

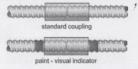
of Stress also reterred to, in general terms, as Y-akit Strength - Iy. (b) For geodechnical applications /5% /5µ may be used for proof resting. Standards: Ø 26.5 - 47mm (grade 950/1050N/mm²) ETA 05/0123 and ETAG 013. Øs 15 & 20mm (grade 900/1100N/mm²) formitie approvals. Øs 57 - 75mm /1035 N/mm²) system approval.

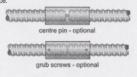
Modulus of Elasticity: E = 205,000 N/mm² +/- 5%. Stock Lengths: $15mm - 20mm\emptyset$ bars, 6.0m; $26.5mm - 75mm\emptyset$ bars, 12.0m. Tolerances +/- 50mm. All bar diameters can be cut to length to suit customer requirements.

Couplers for Threadbars

Couplers enable prestressing steel threadbars to be coupled or extended, reliably and efficiently. Coupler strength (for bar & 26.5.5.47mm) = 1.27 x Yield Strength, which equates to 1.15 x Ultimate Strength, in accordance with German Approval Certificates. Coupler strengths for other prestressing steel bar grades (bar & 15.8 20mm, and 57 - 75mm) exceed the published Ultimate Bar Strengths and are covered by separate approvals (see note C, Technical Data).

Precautions should be taken to ensure that the coupler remains centrally located. This can be achieved through the use of grub screws and/or a centre pin. Marking the two bars with paint or similar at half a coupler length prior to engagement provides visual confirmation of centralisation and is recommended as good working practice.





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Depth	68	ft	Type	Do (in)	t (in)	Di	Astl	plf	Embd.(ft)		L	
fy	80	ksi	1	13.375	0.5	12.375	20.2	68.8	11	79	ft	
f'c	5000	psi	2	13.375	0.5	12.375	20.2	68.8	16	84	ft	
dstl	490	PCF								10		
	0.284	pci	*									
Diar	meter	1A	rea									
75	mm	4418	mm2									
2.95	in	6.848	in2									
	1	RS Me	eans p 29	90								
Dia	Mat'l	Sales	Labor	Equip	O&P				Current	Туре	n	Cost
12	32.00	5%	6.80	4.41	15%					1	2	37722.01
13.375	33.72	5%	7.59	4.92	15%					2	10	200547.39
14	34.5	5%	7.95	5.15	15%					Total	\$	238269.40
		Thick W	all Addit	tion					Proposed	Туре	n	Cost
+	0.93	5%	0	0	15%					1	5	94305.02
	25 5	Reinfo	orcemer	nt						2	12	240656.87
+	0.83	5%	0.34	0	15%					Total	\$	334961.90
	10 O	Final C	ost Figu	res					Diffe	erence	\$	96692.49
Dia	Mat'l	Sales	Labor	Equip	O&P							
13.38	154.84	5%	30.99	4.92	15%							
Loc	1.042		1.670	1.319								
Time					1.01	9						
			Total	\$	238.75							